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# Experimental and numerical study of a top tensioned drilling riser subjected to vessel motion

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# Abstract

Model tests of a top tensioned riser (TTR) model were carried out as a part of a joint industry project, with the purpose of better understanding the dynamic behaviour of drilling riser and verifying the calculations of the riser analysis tools. Sinusoidal motion in one direction was imposed at the top end of the riser model to simulate vessel motion. The tests were carried out in still water, accelerations and bending strains were measured along the riser model. Numerical simulations were performed using RIFLEX and the predicted global responses were compared with the model tests. This paper discusses interesting aspects of this comparison as well as the general dynamic behaviour of the top tensioned riser.

It was found that the dynamic responses of a TTR with vessel motion can consist of not only the IL responses due to vessel motion at the riser top end, but also CF vortex-induced vibrations (VIV) under conditions when Keulegan–Carpenter (KC) number is relatively small. CF VIV response is estimated using a time domain VIV prediction model and compared to the measured response. The main conclusion is that the IL global dynamic responses and CF VIV responses are predicted sufficiently well.

Keywords: Vortex-induced vibrations, Top-tensioned riser, Keulegan–Carpenter number, Time domain

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# 1 1. Introduction

A top tensioned marine riser connects the offshore wellhead (WH) on the seabed and the mobile
offshore drilling unit (MODU) on the free surface,
conveying oil and mud. The marine riser is subject
to dynamic loads caused by waves, currents and
motions of MODU induced by environmental loads

<sup>8</sup> (Yin et al., 2018a). The TTR system is developed
<sup>9</sup> for deepwater drilling and/or workover operations.
<sup>10</sup> TTR is widely deployed by spar or tensioned-leg
<sup>11</sup> platforms (TLP).

VIV of a free-hanging riser due to vessel motion have been investigated by both experimentally and numerically (Jung et al., 2012; Kwon et al., 2015; Wang et al., 2017b,c). Jung et al. (2012) carried out model tests on a scaled free-hanging riser with December 5, 2018

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imposed top oscillations in still water, and studied 52 17 the VIV responses under low KC numbers (2 to 53 18 17). Kwon et al. (2015) studied an Ocean Thermal 19 Energy Conversion (OTEC) riser by conducting IL 20 forced oscillation experiments in designed current 21 on a scaled OTEC riser model. The KC numbers 22 are relatively low (2 to 4). It is found that due to 23 the current, IL VIV responses were weakened, while 24 CF VIV responses were amplified due to larger rela-25 tive velocity. Wang et al. (2017b) performed model test on a free hanging riser with vessel motion in 27 constant current, corresponding KC numbers are 28 between 10 and 80. The out-of-plane VIV responses 29 were observed and the resulted strain was compa-30 rable to the in-plane global responses. Wang et al. 31 (2017c) proposed an empirical prediction method-32 ology to account for vessel motion induced VIV for 33 a free hanging riser under small KC numbers (8.3, 34 12.7).35

Guo et al. (2013) investigated the dynamic re-36 sponses of a TTR under combined excitation of ves-37 sel motion, surface wave and internal solitary wave. 38 The riser is vibrating either at the surface wave fre-39 quency or vessel motion frequency, while the influ-40 ence of internal solitary wave is much larger than 41 the other two excitations. 42

Meng et al. (2017) modelled and simulated a flex-43 ible pipe conveying internal flow in its transition 44 range from being subcritical to supercritical. A 45 combination of internal flow effect and VIV was il-46 lustrated. Distinct different internal flow effect was 47 identified depending on the velocity of the internal 48 flow. 49

Wang et al. (2017a) investigated the VIV of a 50 steel catenary riser (SCR) due to vessel motion, 51

which is equivalent to oscillatory current. The dominant parameter - maximum equivalent current velocity is found to govern the vessel-motion induced VIV.

Shi and Manuel (2017) applied proper orthogonal decomposition (POD) and weighted waveform analvsis (WWA) to the data sequentially to estimate the fatigue damage estimation in an instrumented riser effectively.

Thorsen et al. (2014) has developed a new semiempirical time domain method for VIV prediction. It includes a hydrodynamic excitation force model in which the excitation force synchronizes with the oscillation velocity to obtain lock-in. Thorsen et al. (2016) extended the time domain method by adding a damping formulation, and the excitation force model was optimized by validation against flexible riser VIV tests. The optimized time domain model was used to simulate the CF VIV of an elastic cylinder in oscillating flow at two KC numbers (31 and 178) and maximum reduced velocities. Comparison with experiments shows that the model provides realistic frequency content, dominating mode and amplitude of vibration. Ulveseter et al. (2017)modified the time domain model by modelling the midpoint of the synchronization range as a simplified Gaussian process, and enable it to describe the stochastic stochastic nature of the responses of long slender beams subjected to stationary current.

Yuan et al. (2017) proposed another time domain model which can simulate combined CF and IL VIV. However, the hydrodynamic coefficient in IL direction was taken from pure IL experiments by Aronsen (Aronsen and Larsen, 2007; Aronsen, 2007), instead of combined IL and CF VIV exper-

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iments. The prediction is expected to be improved 122
by using more realistic IL coefficients from com- 123
bined IL and CF VIV experiments, such as Yin 124
et al. (2018b). 125

Significant diverging conclusions on global riser 126 91 analysis were found from different studies. For ex- 127 92 ample, Tognarelli et al. (2008) concludes that 'For 128 93 full scale drilling risers without VIV suppression, 129 94 data show that state-of-the-art analysis methods are, 130 95 on average, inherently 30X conservative on a maxi- 131 96 mum fatigue damage basis.' While after comparing 132 97 global riser analysis of a drilling riser with full scale 133 98 measurement, Grytøyr et al. (2017) concluded that 134 99 'global riser analyses are able to predict the actual 135 100 load levels with reasonable accuracy. However, the 136 101 results actually indicate that there is a slight bias 137 102 towards non-conservative results when studying the 103 square root of sum of squares (SRSS) value of the 104 response, especially for the lower riser response.' 105 The bias is mainly due to the scatter (spreading) in 106 the measured signals, in addition, by adjusting the 107 hydrodynamic coefficients in the prediction tools, <sup>140</sup> 108 the analysis could also be improved. 141 109

With review of the above research works, several 142 110 conclusions could be made: (1) Risers under ves- 143 111 sel motion will not only have the in-plane global 144 112 responses, the out-of-plane VIV responses will also 145 113 be excited; (2) The resultant strain in both direc- 146 114 tions are comparable and should be considered; (3) 147 115 VIV due to vessel motion is equivalent to VIV in 148 116 oscillatory flow, the dominant parameter is the  $KC_{149}$ 117 number; (4) Time domain method is needed to pre- 150 118 dict the vessel motion induced VIV accurately. 119 151

Statoil and BP carried out a comprehensive 152
model test program on drilling risers in MARIN- 153

TEK's Towing Tank in February 2015 (Yin et al., 2018a). The objective was to validate and verify software predictions of drilling riser behaviour under various environmental conditions by using of model test data. The configurations of the model were varied systematically by including different lower boundary conditions, blow-out preventer (BOP) and lower marine riser package (LMRP), buoyancy modules and drill string. In the present paper we only study the top-tensioned bare riser model configuration under forced harmonic motion on the riser top end.

The present study focuses on the dynamic responses of a top-tensioned riser under vessel motions. Part of the results were published in Yin et al. (2018c).

# 2. Theoretical background

#### 2.1. VIV in oscillatory flow

The problem studied in this paper is a toptensioned riser model subjected to sinusoidal motion on the top end in still water, which is equivalent to a drilling riser subjected to oscillatory flow.

The IL hydrodynamic forces include an inertia force and a drag force, the CF hydrodynamic force is the fluctuating lift force due to vortex shedding (Blevins, 1990). When KC > 30, the vortex shedding period  $T_s \approx 5D/\dot{x}_{max}$  is a small fraction of the oscillation period  $T = KCD/\dot{x}_{max}$ , where  $\dot{x}_{max}$  is the maximum oscillating velocity, D is the outer diameter of the riser model. For our case KC < 30, the vortex shedding period is comparable to the oscillating period and strong interaction is expected. After reviewing several earlier experimental 176 work, (Blevins, 1990) classified the vortex shedding patterns of circular cylinders in oscillatory flow, and the relationship between vortex shedding frequency 178 and oscillation frequency, see Tab. 1.

# 159 2.2. Parameters

Several key parameters are discussed and defined by Sumer and Fredsøe (1988) and Blevins (1990). The forced harmonic motion at the top end of the riser x(t) is:

$$\begin{aligned} x(t) &= Asin(\omega t) \\ &= Asin(\frac{2\pi}{T}t) = Asin(2\pi ft) \end{aligned} \tag{1}$$

where A is the oscillation amplitude,  $\omega = 2\pi/T = 2\pi f$  is the angular oscillation frequency.

166 The oscillation velocity  $\dot{x}(t)$  can be derived as:

$$\dot{x}(t) = \omega A\cos(\omega t) = \frac{2\pi}{T} A\cos(\frac{2\pi}{T}t)$$
$$= 2\pi f A\cos(2\pi ft)$$
(2)

Inserting the amplitude of the flow velocity,  $2\pi A/T$ , into the formula for the *KC* number gives:

$$KC = \frac{VT}{L} = \frac{2\pi A}{T} \frac{T}{D} = \frac{2\pi A}{D}$$
(3)

where V is the oscillating velocity, D is the outer diameter of the riser.

<sup>172</sup> The Reynolds number is defined as

$$Re = \frac{\dot{x}(t)D}{\nu} = \frac{2\pi AD}{\nu T} cos(\frac{2\pi}{T}t)$$
$$= \frac{2\pi fAD}{\nu} cos(2\pi ft)$$
(4)

<sup>173</sup> The maximum Reynolds number can be found

$$Re_{max} = \frac{\dot{x}_{max}D}{\nu} = \frac{2\pi AD}{\nu T} = \frac{2\pi fAD}{\nu}$$
(5)

where  $\nu$  is the kinematic viscosity of the fluid.

The reduced velocity  $V_r$  is defined as

$$V_r = \frac{\dot{x}_{max}}{Df_n} = \frac{2\pi Af}{Df_n} = \frac{2\pi A}{D} \frac{f}{f_n} \tag{6}$$

where  $f_n$  is the measured natural oscillation frequency in still water.

### 3. Model test

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The model tests have been performed in the Towing Tank III at MARINTEK (now SINTEF Ocean), see Fig. 1. The Towing Tank III has a dimension of  $L \times B \times D = 85 \ m \times 10.5 \ m \times 10 \ m$ . It is equipped with a double flap wave-maker and a overhead towing carriage. The model tests to be analysed in this paper is carried out in still water.

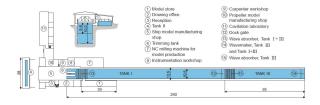


Figure 1: Principle sketch of the towing tank in SINTEF Ocean (earlier MARINTEK).

# 3.1. Test set-up

The general set-up of the model test is illustrated in Fig. 2. The test rig is a steel truss beam which accommodates the drilling riser model. The truss beam is hinged onto the vertical beams on the towing carriage, and it can be lifted to a horizontal position by the overhead crane on the towing carriage when rigging is needed. On the top side, steel substructures are added to enhance the stiffness of the rig and accommodate the horizontal oscillator. On the bottom side of the rig, four chains were spread

- LO	<b>T7</b>	<u> </u>
KC	Vortex pattern	$f_{CF}/f$
< 0.4	No separation.	No CF forces
0.4 - 4	A symmetric pair of vortices is formed in the wake. The vortices reverse during	CF forces are
	the oscillation cycle	minimal.
4 - 8	Asymmetric pair of vortices.	2
8 - 15	Vortex pairs are shed alternately into the wake during each half-cycle of oscil-	
	lation. The vortex pairs convect alternately asymmetrically at $\approx 45 \deg$	
15 - 22	Multiple pairs of vortices are shed per cycle and the pairs convect at 45 deg.	3
22 - 30	Multiple pairs of vortices are shed per cycle.	4
>30	Quasi-steady vortex shedding.	$\approx 0.2 KC$

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Table 1: Vortex shedding pattern and frequency ratio in oscillatory flow, from page 217 of (Blevins, 1990).

diagonally to keep the rig vertical and provide ad-219
ditional stiffness. The drilling riser model is pinned 220
on both ends, and it is pre-tensioned by a compress 221
spring on the top end. Harmonic motion is imposed 222
on the top end by a linear motion system, see Fig. 223
3. The submerged part of the riser model is filled 224
with fresh water. 225

#### 206 3.2. Riser model

The core of the bare riser model was a fibreglass 207 229 reinforced pipe. This core fibreglass pipe has an 208 outer diameter of  $20 \ mm$  and a wall thickness of 209 230 1.5 mm. It was fabricated by a subcontractor, Vello 210 Nordic AS. The optical fibres, accelerometers, and 231 211 their cables were glued on the outer surface of the 232 212 fibreglass pipe. A silicon tube was wrapped around 233 213 the sensors and cables. Due to the cables and sili- 234 214 con tube, the outer diameter of the riser model was 235 215 increased to 28 mm generally. 216 236

At the locations of accelerometers, the outer di- 237 ameter was slightly increased locally at the ac- 238 celerometer locations. When the riser is vibrating, additional structural damping might be introduced by friction between silicon tube and the core fibreglass pipe. The local increase of outer diameter and possible structural damping due to friction were neglected in the numerical calculation.

The properties of the riser model in model scale (MS) and corresponding full scale (FS) values are summarized in Tab. 2. The drilling riser model is in 1:19 scale, and Froude scaling is applied in the present study.

# 3.3. Instrumentation and Data acquisition

The bare riser model was instrumented with fibre optics strain gauges at thirteen (13) locations along the riser. At each location, four fibre optics strain gauges were instrumented, implying 52 strain gages in total. They are used to measure axial stress and biaxial bending stresses. The fibres were glued on the glass fibre rod, in four quadrants of the cross section. The fibres were protected by

Table 2: Riser model properties.

Property	Unit	Model scale	Full scale
Outer diameter, OD	m	0.028	0.532
Inner diameter, ID	m	0.017	0.323
Length, $L$	m	8.996	171
Mass/length, $m/l$	kg/m	0.668	247
Bending stiffness, $EI$	$Nm^2$	120	$3.5 \times 10^8$
Spring stiffness, $K$	N/m	$1.819  imes 10^5$	$6.73  imes 10^7$
Top tension, $T$	N	212	$1.5  imes 10^6$

the outer silicon layer. Additionally, two normal 262 239 strain gauges were instrumented. One is located on 263 240 the bare riser top part, above the water line. One is 264 241 located near the lower end of the riser. Twelve  $(12)_{265}$ 242 two-dimensional accelerometers were instrumented 266 243 on the bare riser. The fibre optic strain signals 267 244 were sampled at a rate of 25 Hz. All other signals 268 245 were sampled at a rate of 200 Hz. Figure 4 presents 269 246 the distribution of accelerometers, fibre optic strain 270 247 gauges, and strain gauges. 248

The displacement is obtained by integrating ac- 271 249 celeration signals measured by accelerometers. The 250 curvature are directly measured by both fibre optic 251 273 strain gauges and normal strain gauges. 252

#### 3.4. Test program 253

The complete test program and other configura- 277 254 tions are described in Yin et al. (2018a). 255

The tests on the top-tensioned riser configuration 279 256 are studied in this paper and listed in Tab. 3. Uni- 280 257 directional sinusoidal oscillation motions were im- 281 258 posed on the top end of the riser model. Eigenvalue 282 259 analysis was performed before the experiments, and 283 260 pluck tests were carried out to verify the eigenfre- 284 261

quencies when the test model was set up. The oscillation frequency was either the  $1^{st}$  eigenfrequency (Test 1015, Test 1020 and Test 1025) or the  $2^{nd}$  eigenfrequency (Test 1005, Test 1010 and Test 1011), in order to excite the first or the second modes. Meanwhile, the two frequencies were considered to be representative for wave frequencies. Oscillation amplitudes were varied under each frequency.

# 4. Numerical simulation

The model test data is used to verify and validate numerical models. The top-tensioned riser model is numerically modelled using RIFLEX (SINTEF Ocean, 2017a). RIFLEX is an efficient program system for hydrodynamic and structural analysis of slender marine structures. It basically includes a finite element module which uses beam or bar elements, and a hydrodynamic loading model described by the generalised Morison's equation. In the present study, RIFLEX simulation is performed under SIMA (SINTEF Ocean, 2017b). SIMA is a workbench developed by SINTEF Ocean (former MARINTEK). It supports the entire process

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Table 3: Test program.

Test	A $[m]$ MS/FS	f [Hz]	<u>vo</u>	$Re_{max}$
No.		MS/FS	KC	
1005	0.026/0.50	1.477/0.399	5.83	$5.93 \times 10^3$
1010	0.052/1.00	1.477/0.399	11.67	$1.18\times 10^4$
1011	0.013/0.25	1.477/0.399	2.92	$2.96\times 10^3$
1015	0.026/0.50	0.646/0.148	5.83	$2.59\times 10^3$
1020	0.052/1.00	0.646/0.148	11.67	$5.19\times10^3$
1025	0.078/1.50	0.646/0.148	17.50	$7.87 \times 10^3$

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from the definition of the simulation and its ex-  $_{\rm 305}$ 285 ecution to the interpretation and documentation 306 286 of the results. The definition of a simulation is  $_{307}$ 287 streamlined through a user-friendly graphical inter-288 308 face with three dimensional visualisation. It offers 309 289 a complete solution for simulation and analysis of  $_{\scriptscriptstyle 310}$ 290 marine operations and floating systems. 291 311

### 292 4.1. Eigenvalue analysis

Eigenvalue analysis is performed to find theeigenfrequencies and corresponding eigenmodes.

#### <sup>295</sup> 4.2. IL responses modelling

The dynamic responses in IL direction is due to <sup>315</sup> the imposed harmonic motion at the top end of the <sup>316</sup> riser model. <sup>317</sup>

Non-linear time domain global dynamic analysis <sup>318</sup> is performed in IL direction (direction of oscillation) <sup>319</sup> to simulate the dynamic responses in in-line direction. Morison's equation (Faltinsen, 1993; SINTEF <sup>321</sup> Ocean, 2017a) is used to calculate the hydrodynamic force in IL direction. <sup>323</sup>

# 4.3. CF responses modelling

To investigate and simulate the VIV in CF direction due to oscillatory flow, a recently developed time domain VIV prediction model is used Thorsen et al. (2014, 2016). This time domain VIV prediction model is implemented in RIFLEX now.

At any point along the riser, the total hydrodynamic force is calculated as:

$$\mathbf{F} = C_M \rho \frac{\pi D^2}{4} \dot{\mathbf{u}}_n - (C_M - 1) \rho \frac{\pi D^2}{4} \ddot{\mathbf{x}}_n + = \frac{1}{2} \rho D C_D |\mathbf{v}_n| \mathbf{v}_n +$$
(7)
$$= \frac{1}{2} \rho D C_v |\mathbf{v}_n| (\mathbf{j}_3 \times \mathbf{v}_n) cos \phi_{exc}$$

The three first terms on the right side of Eq. (7) make up Morison's equation (Faltinsen, 1993), while the final term represents the oscillating lift force due to vortex shedding (SINTEF Ocean, 2017a).  $\rho$  is the water density, D is the outer diameter of the riser.  $C_M$  and  $C_D$  are the inertia and drag coefficients respectively, while  $C_v$  determines the strength of the vortex shedding force. Furthermore,  $\dot{\mathbf{u}}_n$  is the normal component of the fluid particle acceleration which is perpendicular to the cylinder axis,  $\ddot{\mathbf{x}}_n$  is the normal component of the cylinder

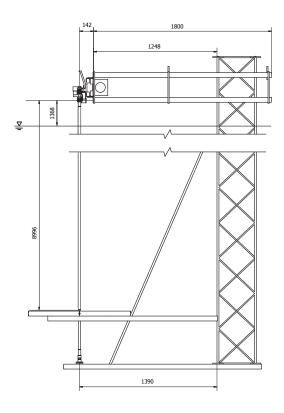


Figure 2: Model test set-up.

acceleration and  $\ddot{\mathbf{v}}_n$  is the normal component of the <sup>343</sup> relative fluid velocity. The relative flow velocity is <sup>344</sup> given as  $\mathbf{v} = \mathbf{u} - \dot{\mathbf{x}}$ , where  $\mathbf{u}$  is the incoming flow <sup>345</sup> velocity and  $\dot{\mathbf{x}}$  is the velocity of the cylinder cross- <sup>346</sup> section.  $\mathbf{j}_3$  is a unit vector pointing in the direction <sup>347</sup> of the cylinder axis.

 $\phi_{exc}$  is a time-varying phase that describes the  $^{348}$  $\phi_{exc}$  is a time-varying phase that describes the  $^{348}$  $\sigma_{331}$  oscillations of the lift force. The evolution in time  $\sigma_{349}$  $\sigma_{332}$  is given by equations (8) and (9):

$$\frac{d\phi_{exc}}{dt} = 2\pi \frac{\hat{f}_{exc}|\mathbf{v}_n|}{D} \tag{8}$$

$$\hat{f}_{exc} = \begin{cases} \hat{f}_0 + (\hat{f}_{max} - \hat{f}_0)sin\theta, \ \theta \ge 0 \\ \hat{f}_0 + (\hat{f}_0 - \hat{f}_{min})sin\theta, \ \theta < 0 \end{cases}$$
(9) 353

Equation (8) gives the relationship between the 355 dimensionless and the actual frequency. Equation 356 (9) models the synchronization between the vortex 357

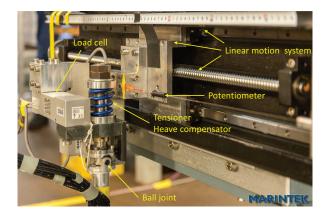


Figure 3: Riser top unit: One degree-of-freedom (DOF) forced motion actuator, tensioner/heave compensator, ball joint, horizontal potentiometer, and three component load cell.

shedding and the cylinder motion. Here,  $\theta$  is the instantaneous phase difference between the cylinder cross-flow velocity and the lift force. The essential feature of the synchronization model is that it is possible for the lift force to vary its instantaneous frequency between  $\hat{f}_{min}$  and  $\hat{f}_{max}$ , and lock on to the frequency of vibration. For more details see Thorsen et al. (2014, 2016).

The empirical parameters used in the present study are given in Tab. 4.

#### 5. Results and discussions

# 5.1. Eigenfrequencies and eigenmodes

The first three normalized eigenvector shapes found from eigenvalue analysis in RIFLEX are shown in Fig. 5. Corresponding calculated eigenfrequencies are compared with measured eigenfrequencies from decay test in still water, see Tab. 5. The difference between calculated and measured eigenfrequencies are within 5%, which is acceptable. This difference might be due to the non-continuity

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Table 4: Empirical parameters used in the hydrodynamic model.

$C_M$	$C_D$	$C_v$	$\hat{f}_0$	$\hat{f}_{min}$	$\hat{f}_{min}$
1.1	1.0	1.3	0.17	0.125	0.3

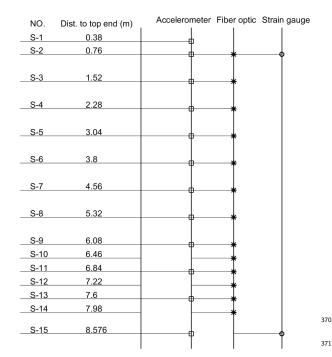


Figure 4: Instrumentation distribution.

of the cross section along the riser, caused by theinstrumentations.

# 360 5.2. Displacement amplitude and orbits

Figure 6 to Fig. 11 show the displacement responses of all tests listed in Tab. 3. In each figure, the plot on the left presents the normalized displacement amplitude along the riser model in both IL and CF directions. The plots on the two right columns show the orbits of twelve (12) cross sections with accelerometers (see Fig. 3).

Figure 6 to Fig. 8 show a combination of  $2^{nd}$  <sub>385</sub> mode of IL response and  $3^{rd}$  mode of CF response.

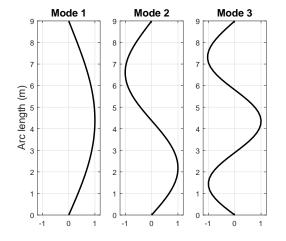


Figure 5: Eigenmodeshapes of displacement.

Fig. 9 to Fig. 11 show a combination of  $1^{st}$  mode of IL response and  $2^{nd}$  mode of CF response. It is expected that test 1005, 1010 and 1011 have the  $2^{nd}$  mode of IL response, while the remaining tests have the  $1^{st}$  mode of IL response. Since it is on purpose to design the tests with imposed top motions with either the  $2^{nd}$  eigenfrequency (1.477 Hz) or the  $1^{st}$  eigenfrequency (0.646 Hz), see Tab. 3. Further discussions will also prove this. It is discovered that the accelerometer in IL direction does not work properly for Test 1015, Test 1020 and Test 1025, it can be seen from Fig. 9 to Fig. 11. The exact reason was unclear, probably because it was not perfectly water-proofed. So the measured signal from this accelerometer are not used for further analysis.

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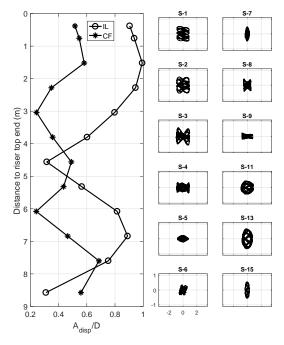
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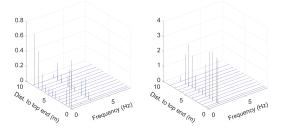
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Table 5: Comparison of eigenfrequencies.

Eigenperiod (Hz)	$f_{n,1}$	$f_{n,2}$	$f_{n,3}$
Model test	0.646	1.477	2.619
Numerical simulation	0.648	1.445	2.503

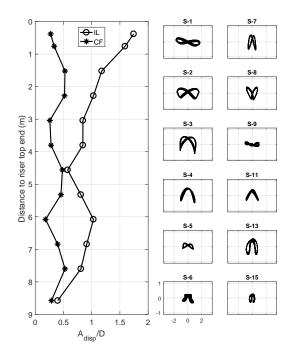


(a) Measured displacement amplitude along the riser model and orbits at locations with accelerometers.

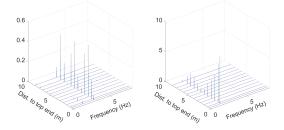


(b) PSD of displacements in CF (left) and IL (right) directions.

Figure 6: Test 1005, A = 0.026 m, T = 0.677 s.

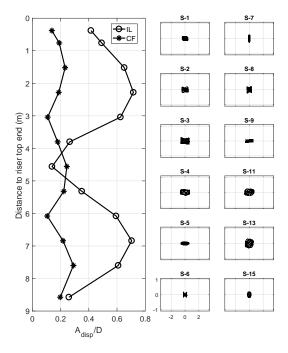


(a) Measured displacement amplitude along the riser model and orbits at locations with accelerometers.

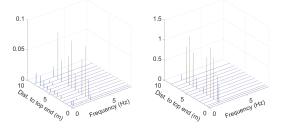


(b) PSD of displacements in CF (left) and IL (right) directions.

Figure 7: Test 1010, A = 0.052 m, T = 0.677 s.

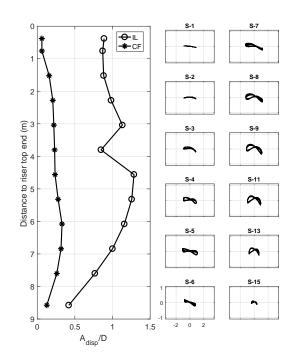


(a) Measured displacement amplitude along the riser model and orbits at locations with accelerometers.

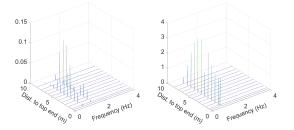


(b) PSD of displacements in CF (left) and IL (right) directions.

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Figure 8: Test 1011, A = 0.013 m, T = 0.677 s.
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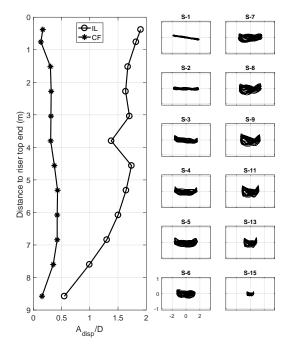


(a) Measured displacement amplitude along the riser model and orbits at locations with accelerometers.

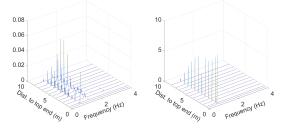


(b) PSD of displacements in CF (left) and IL (right) directions.

Figure 9: Test 1015,  $A = 0.026 \ m, \ T = 1.547 \ s.$ 

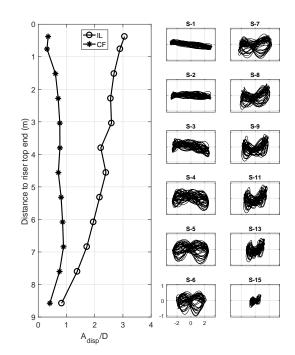


(a) Measured displacement amplitude along the riser model and orbits at locations with accelerometers.

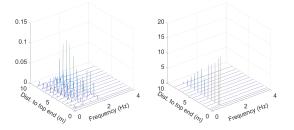


(b) PSD of displacements in CF (left) and IL (right) directions.

Figure 10: Test 1020, A = 0.052 m, T = 1.547 s.



(a) Measured displacement amplitude along the riser model and orbits at locations with accelerometers.



(b) PSD of displacements in CF (left) and IL (right) directions.

Figure 11: Test 1025,  $A = 0.078 \ m, T = 1.547 \ s.$ 

Even though the forced motion is only applied in 417 387 the IL direction, displacements are seen in both IL 418 388 and CF directions. The riser model moves at the  $_{419}$ 389 forced motion frequency in the IL direction. In the 420 390 CF direction, the motion is approximately twice of 421 391 the forced motion frequency, with several other fre- 422 392 quencies additionally. The displacement response 423 393 frequencies are summarized in Tab. 6. Results of 424 394 spectral analysis on the displacement responses are 425 395 shown in Fig. 6b, Fig. 7b, Fig. 8b, Fig. 9b, Fig. 426 396 10b and Fig. 11b. 427 397

The dominating CF response frequency (VIV frequency)  $f_{CF}$  is double of the IL motion frequency  $f_{IL}$  for all six cases. Multiple frequencies in CF displacement responses result in complicated crosssectional oscillation orbits, see Fig. 6a and Fig. 8a. Single frequency displacement responses will give '8-shape' orbits, see Fig. 7a.

The relationship between CF vibration frequency
and the oscillatory flow frequency was defined by
Sumer and Fredsøe (1988):

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$$N = \frac{f_{CF}}{f} = \frac{f_{CF}DKC}{\dot{x}_m}$$
 (10) 440 441 442

where N is the number of vibrations in one cycle of oscillating flow,  $\dot{x}_m$  is the amplitude of the oscillating velocity.

412 If we insert the corresponding values of Test 1010 446

<sup>413</sup> into Eq. 7, we will get N = 2. It is noted that the <sup>414</sup> response pattern for a constant KC number varies <sup>415</sup> with the reduced velocity.

The six tests studied in the present paper have KC number from 5 to 18. In this KC number range, the vortex pattern are asymmetric pair of vortices in one cycle, , see Tab. 1 (Sarpkaya, 1976; Williamson, 1985), and therefore, the frequency ratio  $f_{CF}/f = N = 2$ .

It is important to note that the KC number decreases to zero along the drilling riser from the top end to the bottom end. When KC number is smaller than 4, the force in CF direction is minimal (Blevins, 1990). That explains the multiple frequencies in CF for Test 1005 and 1011, see Tab. 4, Fig. 6a and Fig. 8a. For Test 1010 (Fig. 7a), the maximum KC number is 11.67, vortex pairs are shed alternately into the wake during each half-cycle of oscillation, resulting distinct CF forces which has twice the frequency of IL oscillation, see Tab. 4.

The measurement signals of S-3 of Test 1010 is plotted in Fig. 12, together with the top motion history. In general, all the test cases have relatively low KC number (<20), Test 1010 has a KC number of 11.67, see Tab. 2. We can see that the CF responses are stable, without amplitude modulation. Similar responses were discovered at small KC number in Fu et al. (2014). It is probably due to that at small KC number, the vortex shedding is strengthened by its wake.

#### 5.4. IL response amplitude comparison

IL displacement amplitude comparison is shown in Fig. 13. IL curvature amplitude comparison is

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Table 6: Response frequency.

Test	$V_r$ [-]	$f_{IL} = f  [\text{Hz}]$	$f_{CF}$ [Hz]	$f_{CF,l}$ [Hz]
1005	13.3	1.48	2.77	0.18,  1.48
1010	26.7	1.48	2.95	1.48,  0.22
1011	6.7	1.48	2.67	0.29,  1.48
1015	5.8	0.65	1.29	0.65
1020	11.7	0.65	1.29	0.65
1025	17.5	0.65	1.28	0.65, 1.75, 0.83, 0.45, 0.18

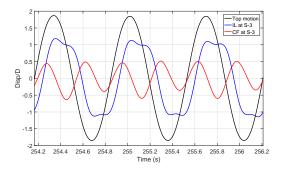


Figure 12: Time history within 3 cycles of Test 1010, at S-3.

shown in Fig. 14. From both figures, the second
mode responses are observed for Test 1005, Test
1010 and Test 1011; while the other three tests
have IL responses dominated by the first mode. RIFLEX simulation over-predict both displacements
and curvatures slightly, which gives conservative estimation.

The curvature amplitude comparison in Fig. 14, 467 454 on the lower part of the riser (6 to 8 m from the 468 455 riser top end), larger differences are observed for the 469 456 first three tests. The experimental measurements 470 457 indicate higher mode curvature may exist in addi- 471 458 tion to the primary mode curvature signal, however, 472 459 RIFLEX seems only capture the dominating mode 473 460 curvature. 461 474

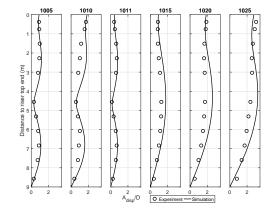


Figure 13: Comparison of IL displacement amplitude.

# 5.5. Cross flow VIV modelling

To study the CF VIV, time domain model developed by Thorsen et al. (2014, 2016) is used. It is based on Morison's equation, with an additional term representing the lift from vortex shedding. The magnitude of the vortex shedding force is given by a dimensionless coefficient,  $C_v$ , and a value of 1.3 is adopted in this study. The drag coefficient has a value of 1.0 in this study. This model allows time varying flow around the structures, it has been validated against some experiments with oscillating flow (Thorsen et al., 2016). The synchronization model within the hydrodynamic load

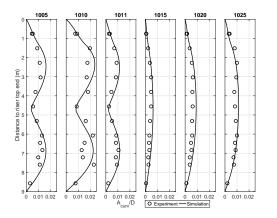


Figure 14: Comparison of IL curvature amplitude.

<sup>475</sup> model is able to capture the vortex shedding pro<sup>476</sup> cess in oscillatory flow. It is important to note that
<sup>477</sup> the present time domain model only predict the CF
<sup>478</sup> VIV responses.

479 Selected results are presented in Fig. 15, Fig. 16480 and Fig. 17.

Figure 15 shows the time history of the IL and CF 481 responses at S-3 together with the top motion, in 482 addition, spectral analysis is shown in the lower plot 506 483 in the same figure. It is observed that dominating 484 frequency of the CF VIV responses is double of the 508 485 IL forced motion frequency, which agrees very well 509 486 with the experimental measurements, see Tab. 4. <sup>510</sup> 487 In addition, strong low frequency component is also <sup>511</sup> 488 observed, which causes that the CF VIV responses <sup>512</sup> 489 have non-zero mean position. The magnitude of <sup>513</sup> 490 the vortex shedding force is proportional to the rel- <sup>514</sup> 491 ative velocity between the oscillating riser model <sup>515</sup> 492 and the flow. A strong vortex will shed when the 516 493 relative velocity is high, and a weak vortex shed- 517 494 ding occur when the relative velocity is low. Since 518 495 the CF frequency is twice the IL frequency, two 519 496 vortices are shed per flow reversal. If one vortex is  $_{520}$ 497

stronger than the other, the mean value taken over 498 one single period will be non-zero. From Tab. 1, 499 we know that asymmetric pair of vortices are shed 500 during the oscillation cycle at the present KC num-501 ber. And over many such cycles you could get a 502 low frequency motion, because the timing is not ex-503 actly equal for each cycle, which means the relative 504 strength of these vortices will vary. 505

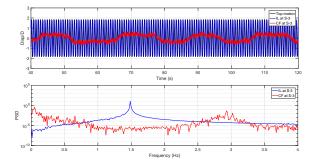


Figure 15: Time domain CF VIV simulation results: time history of top motion and IL and CF motions at S-3 (upper), spectral analysis (lower).

Time history within three cycles of two separate time windows are presented in Fig. 16 and Fig. 17. If we look at the predicted CF VIV responses, Fig. 16 has positive mean value, while Fig. 17 has negative mean value. Moreover, the phase angle between the IL and CF motions are shifted with 180 degrees. Such phase shift was not observed in the experiments. Further studies are needed to investigate whether it is physical or numerical. The predicted amplitude is around 0.6D, which is higher than model test (0.5D), see Fig. 7a.

The comparison of CF VIV amplitude ratio between model test and time domain VIV simulation is shown in Fig. 18. It shows that, using the present input coefficients in Tab. 4 in the time domain VIV

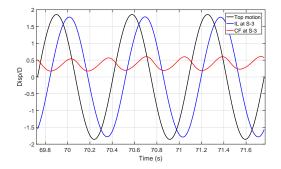


Figure 16: Time domain CF VIV simulation results within 3 cycles of Test 1010, at S-3, selected time window 1.

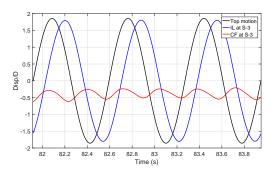


Figure 17: Time domain CF VIV simulation results within 3 cycles of Test 1010, at S-3, selected time window 2.

prediction tool, the CF VIV amplitude is overesti- 538 521 mated, which is conservative. Another observation 539 522 is that time domain VIV prediction does not cap- 540 523 ture the same mode order as the model test. It 541 524 seems that time domain VIV tool predict the  $2^{nd}$  542 525 mode in CF, while model test results show clearly 543 526 the  $3^{rd}$  mode. It is expected that the comparison 544 527 can be improved by optimising the input hydrody- 545 528 namic parameters, but no attempt has been done 546 529 at this stage. On the other hand, the time domain 547 530 VIV prediction tool is semi-empirical, relying on the 548 531 experimental hydrodynamic coefficients. Enriched 549 532 hydrodynamic coefficients database could also im- 550 533 proved the prediction. 551 534

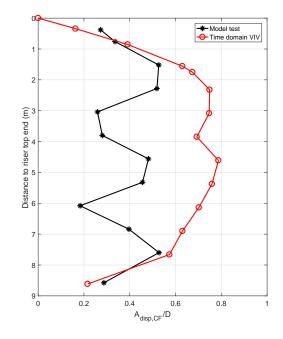


Figure 18: Comparison of CF VIV amplitude ratio along the riser. Model test vs. Time domain CF VIV, Test 1010.

# 6. Conclusions

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A comprehensive drilling riser model test program was performed by a joint industry project funded by Statoil and BP. The model tests were carried out at MARINTEK's towing tank (now SINTEF Ocean) extension in February 2015. Six drilling riser configurations were modelled and Bending strain and accelerations along tested. the drilling riser model in both IL and CF directions were measured by strain gauges, accelerations in both directions were measured by accelerometers. Forces were measured at specific locations. The model tests have simplified but well-defined drilling riser models, covering extensive environmental conditions. The model test data forms a good database, which can be used in many ways, and these help to further understand the compli<sup>552</sup> cate responses of typical drilling risers.

This paper studies the TTR model. One DOF 553 harmonic horizontal forced motions were imposed 586 554 on the top end of the riser model by an actuator. 587 555 The IL responses are induced by the top motion. 588 556 589 Eigen-value analysis and non-linear time domain 557 590 analysis have been carried out by using a riser sys-558 501 tem analysis program RIFLEX. Key results such as 592 559 displacement and curvature amplitudes along the 593 560 594 riser from model tests are compared with the nu-561 merical simulations. Orbits at measurement loca-562 596 tions and spectral analysis results along the riser are 597 563 598 presented in addition. In most of the selected cases, 564 599 RIFLEX over-predicts the displacement and curva-565 600 ture amplitude, indicating conservative prediction. 566 601 Responses in CF direction are measured, which 602 567 603 are caused by VIV due to oscillatory flow. The test 568 604 cases have relative low KC number, the VIV re-569 605 sponses are stable. Even the amplitude of CF VIV 570 606 responses are much smaller than the IL responses, 607 571 608 since the frequency is double as the IL frequency, 572 609 the CF VIV responses may cause significant fatigue 573 610 damage. A recently developed time domain VIV 611 574 prediction tool is applied to simulate the CF VIV 612 575 613 caused by the harmonic IL top motion. The result 576 is promising, the CF VIV frequency is predicted 577 615 correctly. The CF VIV displacement amplitude 616 578 is over-predicted, which will give conservatism in 617 579 618 practice. 580 619

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this study are highly appreciated.

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