

1 **Concept of hydrogen fired gas turbine cycle with exhaust gas**
2 **recirculation: Assessment of combustion and emissions**
3 **performance**

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15
16 Abstract

17 A novel gas turbine cycle concept applicable to power plants with pre-combustion CO₂
18 capture or Integrated Gasification Combined Cycle (IGCC) is presented. These power plants
19 use a hydrogen rich fuel with high reactive combustion properties which makes fuel dilution
20 necessary to achieve low NO_x emissions. The proposed novel gas turbine arrangement is set
21 up as to avoid both fuel dilution and its consequent efficiency penalty, and breakthrough in
22 low NO_x combustion technology. In this concept a high Exhaust Gas Recirculation (EGR)
23 rate is applied in order to generate an oxygen depleted working fluid entering the combustor,
24 enough to reduce the high reactivity of hydrogen rich fuels. As a result the combustion
25 temperature in this environment is inherently limited, thus keeping NO_x formation rate low. A
26 first order assessment of the combustion characteristics under such gas turbine operating
27 conditions is made in the light of a numerical analysis of stability and NO_x emissions
28 potential. Both diffusion and premixed types combustor are considered according to the
29 selected EGR rate. It is first shown that the flame stability could be maintained at EGR rates
30 well above the maximum EGR limit found in conventional natural gas fired gas turbines. The
31 study further shows that at these high EGR rates, considerable reductions in NO_x emissions
32 can be expected. The conclusion of this first order analysis is that there is a true potential in
33 reducing the efficiency penalty induced by diluting the fuel in power plants with pre-
34 combustion CO₂ capture.

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36
37
38 Keywords:

39 Exhaust gas recirculation (EGR), pre-combustion CO₂ capture, IGCC, Hydrogen gas turbine,
40 NO_x

42 1. Introduction

43

44 Hydrogen rich fuels are suitable for gas turbines in three possible applications: (i) the well-
45 established Integrated Gasification Combined Cycle (IGCC) without CO₂ capture; (ii) power
46 plants using the pre-combustion CO₂ capture in the Carbon Capture and Sequestration (CCS)
47 context; (iii) power plants in a fully developed renewable energy based society, where
48 hydrogen is used as energy storage in case of excess wind or solar power. Although CO₂ free,
49 the exhaust gas of a hydrogen fired gas turbine contains pollutants known as Nitrogen Oxides
50 (NO_x) which have been strongly regulated for many decades. During combustion of
51 hydrogen, NO_x formation is mostly controlled by temperature through the thermal NO_x
52 kinetic pathway (also called Zeldovitch'). As the thermal NO_x formation is strongly sensitive
53 to temperature, a small increase in the higher range of temperature results in an exponential
54 increase of NO_x. In fact, NO_x emissions from hydrogen rich fuels have been very well
55 correlated to adiabatic flame temperature both in laboratory scale flame [1] and gas turbine
56 tests [2]. For example, Cocchi et al. [3] were able to model the emissions from a hydrogen
57 fired combustor over a wide range of parameters variation by tuning a model based on the
58 Thermal NO mechanism solely.

59

60 In modern hydrocarbon based gas turbines, the problem of high temperature regions in the
61 flame is avoided by premixing the fuel and air prior to combustion by using lean premixed
62 burners also known as dry low NO_x (DLN) burners. The technology has struggled for many
63 years because the required degree of air – fuel premixing leads to many issues related to
64 combustion stability: flashback, extinction, and thermo-acoustic instabilities [4]. The
65 technology is now commercial and the major gas turbine manufacturers offer engines that
66 achieve NO_x emissions levels within the regulated values without the need of abatement
67 systems (SCR). However, the application of this technology to hydrogen rich fuels still strives
68 because of the specific characteristics of hydrogen combustion: wide flammability limits,
69 much higher reaction rates, preferential diffusion and higher flame temperatures leading to
70 short auto-ignition times and high flame speed [5]. As a consequence, combustion occurs too
71 quickly, before air and fuel have had the time to be adequately premixed, resulting in high
72 temperature and high NO_x emissions. The preferred mode of unwanted flame propagation is
73 flashback through the boundary layer [6, 7], from which the flame dangerously sits in
74 unwanted locations with the risk of component damage. In addition, the flame temperature is
75 higher in hydrogen than in hydrocarbon flames, exacerbating the NO_x formation issue.

76

77 To date the solutions to lower NO_x emissions to acceptable levels are expensive in terms of
78 efficiency penalties or OPEX/CAPEX of end of pipe technologies as for example Selective
79 Catalytic Reduction (SCR) [8]. Considerable development has been made for the syngas fired
80 gas turbine of conventional IGCC plants where hydrogen is the major fuel component and
81 commercial plants are available. IGCC plants with pre-combustion CO₂ capture operate
82 similarly to the plants without CO₂ capture, but with the inclusion of a water gas shift reactor
83 and a CO₂ separation unit upstream the power island which is thus fired with high content
84 hydrogen fuel (cf. Table 2). With or without CO₂ capture, the NO_x formation problem in the
85 diffusion type combustor is tackled by using large amounts of diluent in the high hydrogen

86 content fuel. Nitrogen and steam are both potential diluent candidates because they are
87 available at relatively low cost on site of IGCC plants. Steam/fuel ratio of unity was shown to
88 half the NO_x emissions from 800 ppm @ 15% O₂ dry (1.6 g/Nm³) in Sigali et al. [9, 10].
89 Although steam is demonstrated to be more effective than nitrogen [11], the latter is preferred
90 firstly because steam affects significantly the heat transfer properties of the hot exhaust gas
91 flow and reduce components life [5, 12]. Secondly, nitrogen is a readily available by-product
92 of the Air Separation Unit (ASU) present on site for producing O₂ for the gasifier.

93
94 Good emissions results have been proven in industrial cases with syngas and the use of
95 diluents on diffusion type combustors as reported in several works [2, 11, 12]. Although
96 available at low costs, using nitrogen as diluent induces an expense of up to 20% to 30% of
97 the total auxiliary power consumption required for its compression to slightly above cycle
98 pressure. For comparison this share is even higher than that of the CO₂ compression power in
99 the case of pre-combustion plant [13]. From a cost perspective the compressor unit is
100 expensive and bulky. Gazzani et al. [12] showed that dilution used in combination with
101 diffusion type combustors imposes an efficiency penalty of 1.5 percentage points as compared
102 to the reference combined cycle plant if the amount of nitrogen dilution is that required to
103 reach a flame temperature similar to that of a natural gas flame. The penalty becomes 3.5
104 percentage points in the case of steam dilution. The selected dilution degree and
105 corresponding efficiency decrease is to be compromised with NO_x emissions since these are
106 exponentially proportional to combustion temperature [1, 2].

107
108 The implementation of DLN combustors would avoid the inert dilution to lower NO_x
109 emissions. However, to counteract the aforementioned excessive flashback propensity, high
110 injection velocity and therefore high pressure drop would be needed, which in turns has an
111 efficiency cost as shown in Gazzani et al. [12]. Consequently, DLN burners have not been
112 achieved to date for high hydrogen content fuels. Note that even if lean premixed combustion
113 (i.e. low temperature) of hydrogen could be achieved through DLN burners, Therkelsen et al.
114 [14] measured NO_x emissions that were still higher than in a methane flame at the same
115 temperature. They attributed this effect to the higher propensity of the H₂ – air chemical
116 kinetic to produce NO through the low temperature NNH pathway [15, 16].

117
118 The present work suggests a gas turbine cycle concept that has a potential for low NO_x
119 emissions without the need of either fuel dilution or combustion technology breakthrough. By
120 recirculating the exhaust gas to the gas turbine compressor inlet, the air entering the
121 combustor is oxygen depleted, and inherently limits the combustion temperature and NO_x
122 formation. With this concept, the burner and combustor are simple and reliable (diffusion
123 type) and would avoid the high cost and risks associated with the development of complex
124 DLN burners and combustor arrangements for hydrogen rich fuels. The concept is already
125 known within conventional natural gas combined cycles (NGCC) as Exhaust Gas
126 Recirculation (EGR) [17-19], but for power cycles based on hydrogen fuels, it has to our
127 knowledge, not been evaluated in the scientific literature, apart from a preliminary study by
128 the authors [20]. The study aims at assessing the combustion properties and NO_x emissions at

129 various EGR rates to assess the technical feasibility of such concept in terms of combustion
130 stability and emissions.
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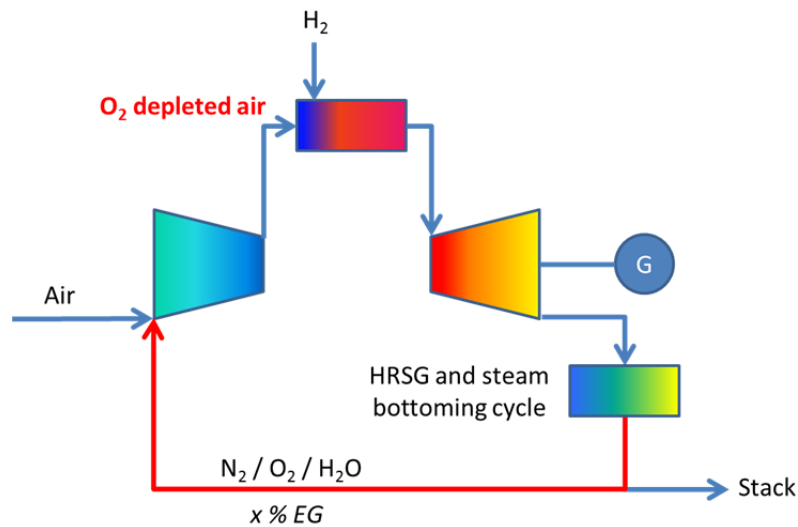


Figure 1: Simplified layout of the hydrogen fired gas turbine with exhaust gas recirculation concept.

132

133 2. Description of the hydrogen fired gas turbine with EGR

134 2.1. Power cycle concept

135 The proposed core gas turbine cycle is depicted in Figure 1. The turbine exhaust gas of a
136 hydrogen fired gas turbine is composed of mostly nitrogen originating from the air, steam
137 being the product of hydrogen combustion, excess oxygen and minor fractions of carbon
138 dioxide. The basis of the concept is to adapt Exhaust Gas Recirculation (EGR) to the cycle,
139 where the EGR rate is defined as the ratio of the volume flow of recirculated exhaust gas to
140 that of exhaust gas. By recirculating a fraction of the turbine exhaust gas back to the gas
141 turbine compressor inlet, the gas flow through the compressor and entering the combustor has
142 a reduced oxygen concentration. The NO_x formed by the combustion of hydrogen in O₂
143 depleted atmosphere, is intrinsically limited by the reduced achievable adiabatic temperature.
144 With a conventional fuel like natural gas or oil, the potential of such technique would rapidly
145 be limited by flame stability [21]. However the very reactive characteristics of hydrogen as
146 fuel circumvent this shortcoming as it will be demonstrated in this study.

147

148 The moisture content of the turbine exhaust gas to be recirculated can be controlled through
149 condensation before recycling into the inlet of the gas turbine. On the one hand steam has the
150 positive effects of increasing the total mass flow and reducing the NO_x formation. On the
151 other hand high moisture concentration can lead to problems on the hot turbomachinery parts
152 such as higher heat transfer to the turbine inlet blades (corrosion, thermal barrier coating
153 degradation), and also on the compressor (corrosion, fouling). EGR in gas turbines is also the
154 general principle used in semi-closed oxy-fuel gas turbine cycles where the goal is to replace

155 all the air by the products of combustion of a pure oxygen fired combustor. Thus the working
 156 fluid becomes CO₂ which has very different physical properties than air, implying the need
 157 for a total re-design of the power cycle layout and components such as turbomachinery [22].
 158 When EGR is applied to a hydrogen fired Brayton cycle, the working fluid is also affected to
 159 become richer in nitrogen and in the case of wet EGR, also richer in water vapour content.
 160 The effect on the physical properties of the working is shown for an extreme recirculation rate
 161 of 60% in Table 1, showing that these are nearly identical and very close to the air case in dry
 162 and wet EGR modes respectively. Indeed, close molecular weights and specific heats ratios
 163 imply similar mass flow and isentropic efficiency, hence unaltered thermodynamic
 164 performance for the gas turbine. The impact can therefore be expected to be less than in
 165 conventional natural gas fired gas turbine with EGR as described in Li et al. [17].
 166

167 Table 1: Working fluid properties at selected EGR rate at 288 K compressor inlet temperature.

Case	MW (g/mol)	C _p /C _v (-)	Sound speed (m/s)	T _{comp. exit} * (K)
Air	28.86	1.401	340.9	689
60% dry EGR	28.83	1.400	341.0	688
60% wet EGR	26.55	1.384	353.2	671

168 * Polytropic efficiency=0.95, pressure ratio=18.1 bar as in [13].

169 2.2. Combustion concept

170 In a conventional gas turbine combustor, part of the air is drawn into the primary flame zone
 171 to ensure flame stability. The remaining air is further split where a fraction is used for liner
 172 cooling purposes and another for dilution of the combustion products in order to reach the
 173 turbine inlet temperature and homogenise the temperature profile of the flow entering the
 174 turbine stage [23]. In modern DLN combustors, a higher part of the air is used in the primary
 175 zone in order to be premixed with the fuel such as to limit the maximum flame temperature.
 176 Most manufacturers have tackled the problems related to flashback and stability in this
 177 manner, and the technology is now commercial for hydrocarbon fuels. For the reasons
 178 explained previously, a different alternative to the DLN burners based on lean premix
 179 technology is necessary for hydrogen rich fuels due to its high reactivity.
 180

181 In the present concept, the mixture of oxidizer and H₂ fuel is kept less reactive by depleting
 182 the oxygen in the air through recycling of the exhaust gas. Elkady et al. [21] in a NG fired gas
 183 turbine could operate a gas turbine combustor with up to 35% recycling of turbine exhaust
 184 gas, and an O₂ concentration of 17.8%. Ditaranto et al. [24] has shown in a laboratory scale
 185 swirl stabilized burner that combustion of methane could be sustained in a exhaust gas of a
 186 gas turbine at O₂ concentration levels as low as 15%. It is expected that this limit can be
 187 further reduced with hydrogen thanks to its higher reactivity.
 188

189 Ideally a diffusion combustor would be used due to its design simplicity and low cost, and
 190 without inert gas dilution of the fuel for efficiency loss reasons as explained previously. The
 191 EGR concept has the potential for fulfilling both these requirements. Nevertheless, the

192 reactivity of hydrogen, which limits the implementation of premixed burner technology, is no
193 longer a barrier when the air is sufficiently oxygen depleted. Therefore the present concept
194 based on EGR also enables the use of premixed burners with hydrogen combustion and push
195 the potential for low NO_x performance further down.

196 3. Methodology

197 The expected combustion characteristics in terms of stability and NO_x emissions are
198 evaluated through kinetic modelling of a premixed freely propagating flame using the
199 chemical kinetic code LOGEsoft [25]. The combustion stability is estimated based on the
200 comparison of the laminar flame speed of the H₂ rich fuel in EGR conditions against the value
201 in a conventional known combustor configuration. The approach has been used in Sundkvist
202 et al. [22] and this study uses the same combustor as reference. For predicting the NO_x
203 emissions the calculated concentrations from equilibrium chemical kinetics are compared to
204 the value of that of a H₂ flame in air in the same pressure and temperature conditions. The
205 absolute values obtained in a laminar premixed flame are surely different than what could be
206 expected in a turbulent flame developing in an industrial combustor, but in this work the
207 analysis is focused on the relative reduction obtained which is purely controlled by chemistry.
208 This approach must be considered as a first order evaluation of stability and NO_x emissions as
209 it does not include the complexity of coupled turbulence – chemistry interaction, but it gives
210 relevant trends and sets the limits of feasibility study of the power process concept. The
211 results of this study further define the operational boundaries of the gas turbine engine and
212 particularly that of the combustion unit.

213
214 The fuel and working fluid composition and temperature at the combustor inlet are calculated
215 from the gas turbine case of the EBTF Guidelines IGCC cycle [13]. The fuel definition is
216 given in Table 2. The exhaust gas of the gas turbine is recycled at various rates and the
217 composition is re-calculated in each case. The combustor inlet temperature is calculated by
218 assuming a 0.95 polytropic efficiency at the compressor stage and a constant pressure ratio of
219 18.1 bar. The reaction mechanism used in the premixed flame calculations is the full GRI-
220 Mech 3.0 mechanism including the NO_x subset, which in total contains 53 species and 325
221 reactions [26].

222
223 Table 2: Fuel composition (% vol.) [13].

H ₂	CO	CO ₂	N ₂	Ar	H ₂ O
85,64%	2,66%	3,24%	7,27%	1,14%	0,05%

224 4. Results and discussion

225 4.1. Diffusion combustor mode

226
227 The first generation gas turbine combustors were fitted with safe and simple burner, so-called
228 diffusion type, where the non-premixed fuel and air are injected directly into the primary

229 combustion zone. The main duty of such a combustor was to ensure good ignition and
 230 stability of flame. This type of combustor was rapidly obsolete when emission limits on nitric
 231 oxides (NO_x) became more and more stringent. In diffusion combustion, reactions develop at
 232 the air and fuel interface where reactants have mixed enough to reach flammability limits. As
 233 a result the flame sits preferentially at the near stoichiometric locations where the reaction
 234 rates are higher. This is also the location where the temperatures are closer to the maximum
 235 flame adiabatic temperature and therefore the location of highest NO formation as the thermal
 236 NO pathway is strongly temperature dependent [27]. In hydrogen flames, once NO is formed
 237 it cannot be reduced as there are no CH_i species that can activate the NO reburning
 238 mechanism. As a consequence, even if the global air to fuel ratio is large, NO_x emissions
 239 from diffusion flame burners must be assessed by estimating the NO_x formed in
 240 stoichiometric conditions. In this section, calculations in stoichiometric proportion of fuel and
 241 air are made in order to evaluate the potential of NO reduction by applying EGR to the
 242 hydrogen fired gas turbine.
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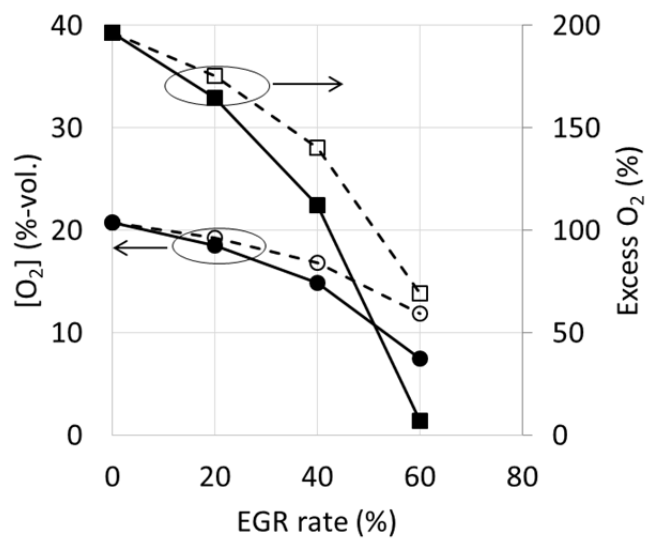


Figure 2: Oxygen concentration in the working fluid entering the combustor and corresponding excess oxygen over stoichiometric combustion value. Filled symbols: wet EGR; open symbols: dry EGR.

244 Recirculating exhaust gas depletes the combustion air of O₂ by dilution with H₂O and N₂ as
 245 shown in Figure 2. Oxygen concentration drops rapidly as the EGR rate increases and faster
 246 in wet than in dry mode. At some point the oxygen concentration is so low that an under-
 247 stoichiometric amount is reached as shown in Figure 2. The adiabatic combustion temperature
 248 in stoichiometric combustion shown in Figure 3 is the highest temperature that can be found
 249 in diffusion flames. At 60% EGR rates it steeply decreases from 2750 K down to 1580 K and
 250 2020 K wet and dry respectively. As the thermal NO_x pathway has a strong exponential
 251 dependency to temperature, the accompanying decrease in NO_x formation as EGR increases
 252 shown in Figure 3 is very effective. Indeed, NO_x concentrations are halved for an EGR rate of
 253 ca. 30% and 40% in wet and dry EGR modes respectively as compared to the case without
 254 EGR. These equilibrium calculations are more qualitative than quantitative as several
 255

256 parameters contributing to the NOx concentration such as residence time in the high
 257 temperature zone (hardware dependent), turbulence – chemistry interactions (burner design
 258 dependent), and radiative heat loss effects are not taken into account. Nevertheless these
 259 results are useful in predicting the trends in a worst case scenario and they indicate that a
 260 moderate rate of recirculation could achieve strong reduction in NOx without the expense of
 261 nitrogen dilution and compression as in the reference case [12, 13]. Cocchi et al. [9] measured
 262 a three-fold increase of NOx emissions when switching from natural gas to hydrogen fuel in a
 263 diffusion type gas turbine combustor. According to Figure 3 a 40% wet or 50% dry EGR rate
 264 would compensate equivalently such an increase in NOx emissions.
 265

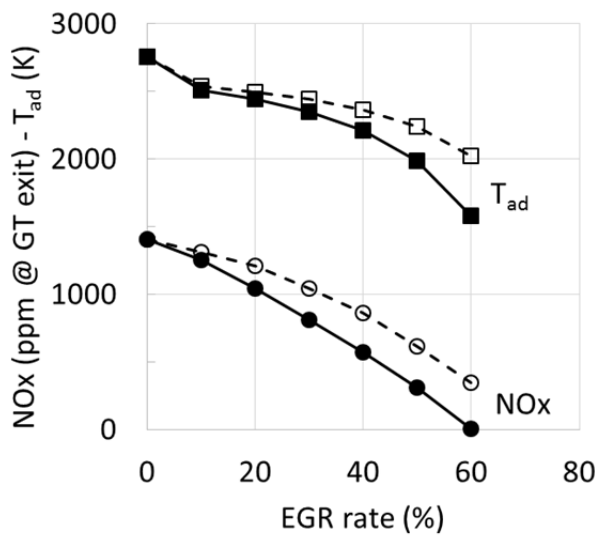


Figure 3: Calculated equilibrium NOx corrected at the GT exit and adiabatic temperature of stoichiometric combustion of the hydrogen rich fuel vs. EGR rate at gas turbine conditions. Filled symbols: wet EGR; open symbols: dry EGR.

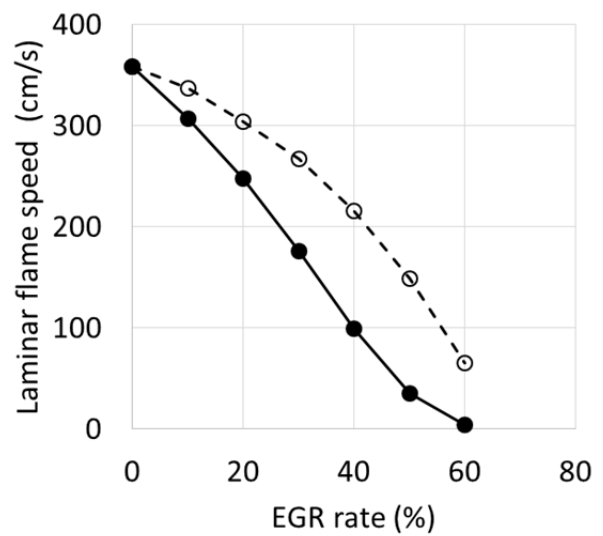


Figure 4: Laminar flame speed of stoichiometric combustion of the hydrogen rich fuel vs. EGR rate at gas turbine conditions. Filled symbols: wet EGR; open symbols: dry EGR.

266
 267 Flame stability and NOx have been strongly related topics in clean combustor development in
 268 the last decades. The general strategy for lowering NOx production has been to tackle the
 269 high temperature regions of the flame by premixing air and fuel in large ratio. The further
 270 away from stoichiometry the premixing ratio is, the lower the flame temperature is, but
 271 unfortunately the weaker the combustion stability is too. Although flame stability in industrial
 272 burner is driven by a complex combination of aerodynamics and chemistry, laminar flame
 273 speed is a good first order indicator of flame stability. The laminar flame speed is a
 274 combustion property for a given mixture determined through kinetic chemistry. Figure 4
 275 shows that at EGR rates of up to 50% and 60% for wet and dry EGR respectively, laminar
 276 flame speeds are close to 50 cm/s. As a reference, gas turbine combustor fired with natural
 277 gas have laminar flame speed values in the order of 20 cm/s, suggesting that stability should
 278 not be impaired by these high levels of EGR. This is due to the well documented positive
 279 effect of hydrogen as fuel on different configurations such as jet flames in co-flow [28],
 280 counter-flow flames [29], and swirl stabilized flames [30] to name but a few. The EGR rate

281 limit in gas turbine with conventional fuels was identified by ElKady et al. [21] at around 35
282 % after which stability issues started to arise, but with hydrogen this limit is pushed further
283 higher. In that concept the reactivity problem of hydrogen is then turned into a benefit by
284 increasing the potential for higher EGR rates and stronger temperature decrease. The belief in
285 the concept is further strengthened by the tests done in a full scale single burner in York et al.
286 [31], where a premixed burner fuelled with 66% H₂ and 34% N₂ at 17 bar showed a reduction
287 of NO_x emissions with a 20% N₂ diluted air.

288 5.2. Premixed combustor mode

289 In conventional gas turbine combustors, part of the air is drawn into the primary flame zone
290 (PZ) where most of the heat release occurs, while the remaining air is used for liner cooling
291 purposes [23]. In modern gas turbines, more and more air is used in the PZ to fulfil the lean
292 requirement of DLN burners. The air bypassing the PZ is introduced as dilution of the flame
293 products to reach the maximum allowable temperature at the turbine stage. The degree of
294 distribution between the primary and dilution air is design strategy, hence manufacturer
295 dependent. By using a premixed flame configuration, we consider in this section the potential
296 of NO_x emissions reduction in the case where a perfect premixing is achieved in the PZ. As
297 discussed previously, premixed burners are challenging if not impossible to achieve with
298 hydrogen, and therefore require a certain degree of EGR to be applicable. Laminar flame
299 speeds are presented together with NO_x calculations in order to assess at which EGR levels
300 premixing is achievable.

301
302 For the EGR gas turbine concept, a combustor strategy needs to be chosen to split the working
303 fluid now composed of air and recycled exhaust gas. The conservative guideline used in this
304 first approach is to maintain the PZ temperature and laminar flame speed close to those of a
305 conventional combustor PZ. The reference combustor used has a PZ adiabatic temperature of
306 1780 K and a turbine inlet temperature (TIT) of 1583 K as in Sundkvist et al. [22]. The
307 laminar flame speed at the reference combustor PZ conditions is in the range 16 – 20 cm/s.
308 Results with working fluid distribution strategies from 45% to 100% in PZ are shown in
309 Figure 5 to Figure 8. NO_x emissions shown with different air distributions in the PZ are given
310 at the gas turbine exit station. In other words, NO_x concentration is calculated in the PZ
311 conditions and then diluted with the bypassed working fluid. This approach is acceptable as
312 the NO chemistry is little active at the lower temperature. NO_x results in Figure 5 and Figure
313 6 are normalized by a reference NO_x value calculated at stoichiometric conditions with no
314 EGR, in order to compare with the diffusion flame case.

315
316 As expected, the more working fluid in the flame zone, the lower the NO_x emissions due to
317 leaner equivalence ratio. At 45% in PZ the conditions are close to stoichiometry, which means
318 high temperature for low EGR rate cases. At this low distribution of working fluid in the PZ,
319 EGR rate is limited by the amount of oxygen available in the PZ. At 40% wet EGR (Figure 5)
320 the PZ is under-stoichiometric, explaining why NO_x drops suddenly to nearly zero. Operating
321 a combustor with a fuel rich primary zone is not impossible and can be seen as a relevant
322 technology (a.k.a. Rich Quench Lean combustor), particularly with hydrogen rich fuels which
323 produce stable combustion and do not exhibit eventual problems of unburned hydrocarbons

324 and soot. The analysis of a detailed combustor layout for the EGR concept is however outside
 325 the scope of this study and we conservatively consider a PZ with 45% of working fluid as the
 326 lower limit of possible split ratio.
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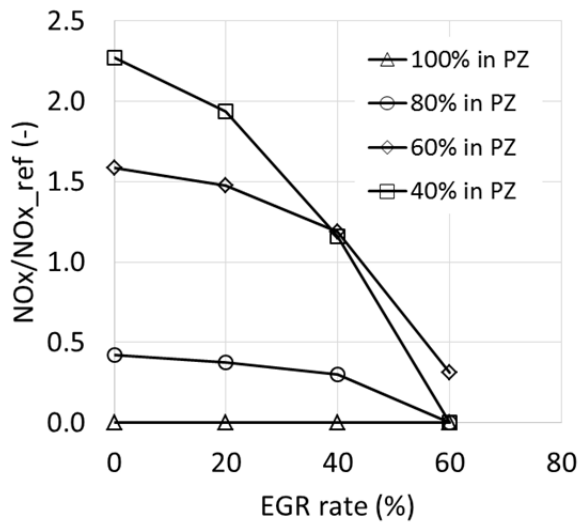


Figure 5: Calculated equilibrium NOx emissions from the gas turbine with dry EGR at different working fluid distribution ratios in the primary flame zone (PZ) of the combustor.

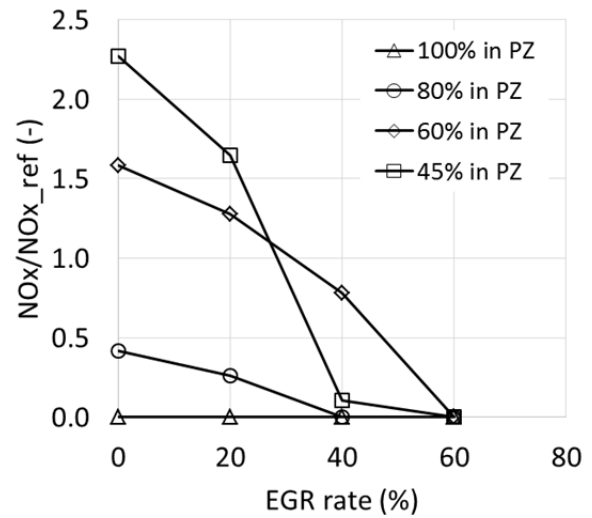


Figure 6: Calculated equilibrium NOx emissions from the gas turbine with wet EGR at different working fluid distribution ratios in the primary flame zone (PZ) of the combustor.

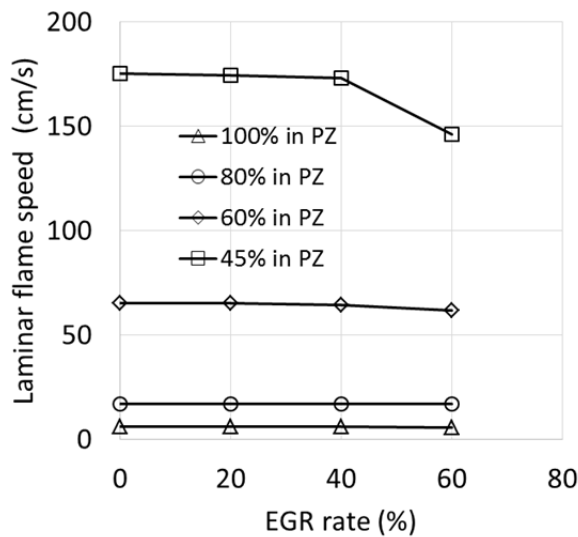


Figure 7: Laminar flame speed property in the primary flame zone (PZ) of the combustor with dry EGR at different working fluid distribution ratios.

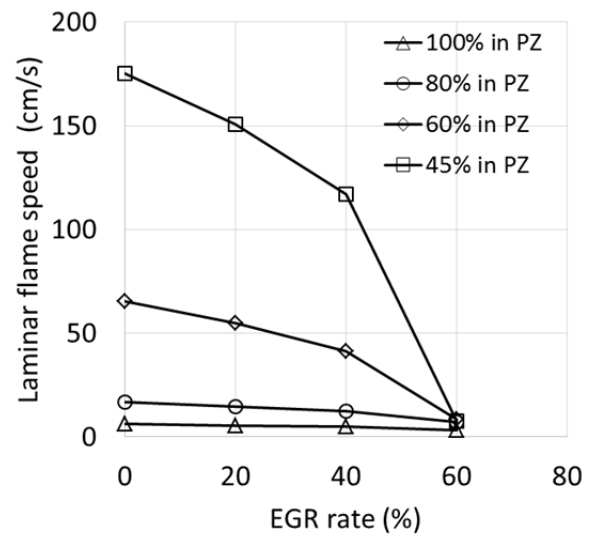


Figure 8: Laminar flame speed property in the primary flame zone (PZ) of the combustor with wet EGR at different working fluid distribution ratios.

328
 329 An interesting finding is that the combustion temperature at constant working fluid
 330 distribution in PZ and varying dry EGR rate is almost constant, indicating that the NOx
 331 reduction observed with EGR is only achieved through a kinetic effect independent of
 332 temperature. This kinetic effect is controlled by the availability of oxygen which
 333 concentration decreases as EGR increases. The NOx reduction in the case of wet EGR seen in

334 Figure 6 is stronger than in dry EGR because of the presence of steam which has a double
335 effect. First, there is a kinetic effect on the NO chemistry through changes in the pool of
336 radicals induced by the increase in H₂O. Secondly the adiabatic temperature is reduced due to
337 the higher heat capacity of steam as compared to the other species it replaces, namely N₂ and
338 O₂. It can be observed from Figure 7 that for a given PZ split, the laminar flame speed for dry
339 recycle has a very little dependency on the EGR rate. This singularity is also linked to the
340 previous observation that the temperature remains constant as dry EGR rate varies.

341

342 Bearing in mind that the laminar flame speed of the reference combustor is in the range of 20
343 cm/s, the results in Figure 7 and Figure 8 indicate that working fluid up to 80 % in the PZ is
344 possible, with corresponding low NO_x potential. However, to achieve premixing before
345 combustion the laminar flame speed at stoichiometric conditions must be low enough. The
346 values given in Figure 4 indicate that high enough EGR rates would be necessary to avoid
347 potential flashback that would render impossible DLN technologies.

348 5.3. Practical application

349 Figure 9 is an attempt to map the boundaries of applicability of the DLN and diffusion
350 combustor technologies as a function of EGR rate in the dry and wet modes. The charts show
351 that achievable premixed technology in dry mode is only possible within a restricted working
352 fluid distribution, and acceptable NO_x emissions require an EGR rate of approximately 50 %
353 depending on PZ distribution. For the wet case, there is a larger possible premixed flame
354 domain, but limited to higher EGR rates if low NO_x emissions are to be achieved. The
355 diffusion combustor technology with ensured low NO_x has a much wider range of
356 applicability as long as EGR rates are above ca. 50 %. We recognize that these limits are quite
357 crude because NO_x is calculated on the basis of equilibrium calculations and the stability is
358 assessed through a simplified manner. Nevertheless they indicate the feasibility of the concept
359 and the design difference that can be expected with regards to existing technologies where
360 typically more than 80% of the air is drawn into the primary zone of current DLN combustors.

361

362 From a first order combustion assessment, the concept seems promising and issues related to
363 the power plant integration must be evaluated, such as efficiency gain or loss, optimization of
364 the plant arrangement and the impact of parameters like the recycle rate. Indeed, the
365 application of the EGR principle will affect the bottoming cycle which is very sensitive to the
366 turbine exhaust gas temperature and mass flow. Proper integration is therefore necessary and
367 there are different options that are conceivable. For example, to reduce the recirculation rate
368 and the cooling demand in the condenser, the available nitrogen which comes as a free stream
369 during the oxygen separation in the ASU, can be injected in the fresh air entering the
370 compressor and thus reduce the amount of recirculation rate.

371

372 The study presented here is based on a pre-combustion CO₂ capture case with a fuel that has a
373 85 % vol concentration of hydrogen (cf. Table 2), however the concept is applicable to a
374 conventional IGCC plant without capture. A syngas fuel with a lower H₂ concentration, but
375 higher CO concentration has similar challenges since the combustion temperature of CO and
376 H₂ are very close.

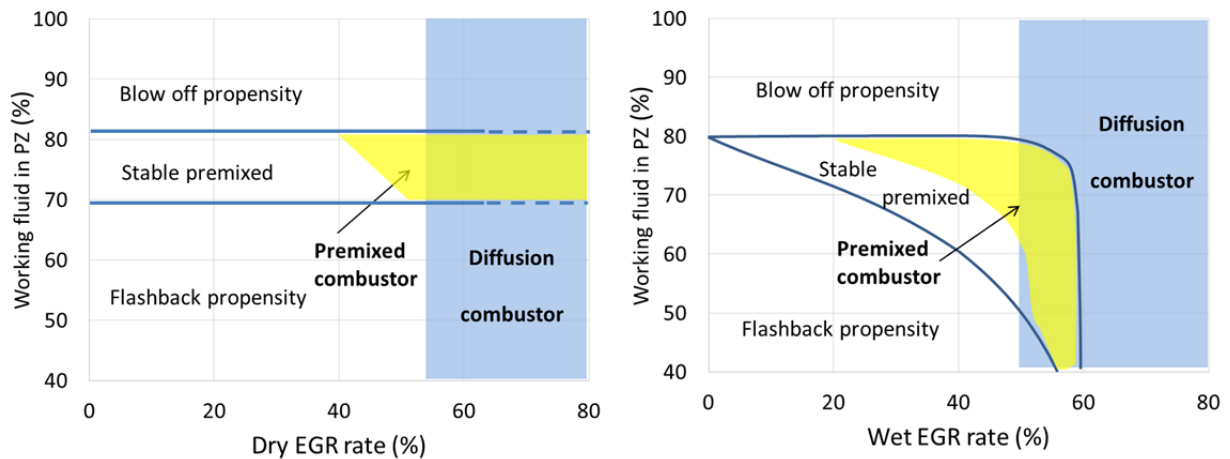


Figure 9: Selection of combustor technologies for the wet (LHS) and dry (RHS) EGR concept for low NO_x emissions hydrogen fired gas turbine. Bold lines represent the limits of flame stability in premix flame mode; the shaded yellow area represents the operation island where premix mode is achievable and within low NO_x limits; the shaded blue area represents the diffusion mode within low NO_x limits.

378 5. Conclusions

379 A novel gas turbine cycle concept for power plant with pre-combustion CO₂ capture or IGCC
 380 is presented. Large inert gas dilution of the hydrogen rich fuel is commonly used to achieve
 381 low NO_x emissions due to the high temperature combustion properties of hydrogen. The
 382 proposed gas turbine arrangement is set up to avoid the efficiency penalty associated with the
 383 dilution by applying high Exhaust Gas Recirculation (EGR) rate to generate an oxygen
 384 depleted working fluid. In this study, a first order assessment of the combustion
 385 characteristics in such a gas turbine condition is made and showed that with an oxygen
 386 depleted oxidizer, the high reactivity of hydrogen fuels is turned into a benefit to potentially
 387 achieve low NO_x emissions. The conclusions on the combustion behaviour in such a cycle are
 388 as follows:

- 389
- 390 1. At high EGR rates the working fluid is so oxygen depleted that stoichiometric flame
 391 temperature are maintained low enough to avoid high NO_x formation.
- 392 2. Flame stability can be maintained at high EGR rates because laminar flame speeds are
 393 high enough thanks to the high reactivity of hydrogen.
- 394 3. Increasing EGR rate in dry mode reduces NO_x formation only through the kinetic
 395 effect of lower O₂ availability.
- 396 4. Increasing EGR rate in wet mode reduces NO_x formation stronger than in dry mode
 397 because a thermal effect driven by increased heat capacity adds to the kinetic effect.
- 398 5. Diffusion combustors could be used, but at high enough EGR rates (i.e. very O₂
 399 depleted working fluid) the use of lean premixed burners becomes also feasible thanks
 400 to the reduced reactivity of hydrogen.
- 401 6. Dry EGR is possibly the most efficient way of abating NO_x because of the good
 402 overlap between stable operating flame conditions and low NO_x formation regions.

403

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410 **References**

- 411
- 412 1. Ströhle J., H.J., Seljeskog M., Ditaranto M., Langørgen Ø., Jakobsen J., Rønnekleiv
413 M. *Experimental and numerical investigation of NOx emission characteristics of*
414 *swirled hydrogen rich flames.* in *8th International Conference on Greenhouse Gas*
415 *Control Technologies (GHGT-8).* 2006. Trondheim, Norway: Elsevier.
 - 416 2. Todd, D.M., Battista, R. A. *Demonstrated Applicability of Hydrogen Fuel for Gas*
417 *Turbines.* in *Proc. of the IchemE Gasification 4 Conference.* 2000. Noordwijk, The
418 Netherlands.
 - 419 3. Cocchi S., G.N., Provenzale M., Zucca A., Romano C., Ceccherini G. , *A Simple*
420 *Model for NOx Formation in Diffusion Gas Turbine Combustors: Rig Test Validation*
421 *with a Wide Range of Fuel Gases,* in *31st Meeting on Combustion,* I.S.o.t.C. Institute,
422 Editor. 2008: Torino, Italy.
 - 423 4. Candel, S., *Combustion dynamics and control: Progress and challenges.* Proceedings
424 of the Combustion Institute, 2002. **29**: p. 1-28.
 - 425 5. Chiesa, P., G. Lozza, and L. Mazzocchi, *Using hydrogen as gas turbine fuel.* Journal
426 of Engineering for Gas Turbines and Power-Transactions of the Asme, 2005. **127**(1):
427 p. 73-80.
 - 428 6. Lin, Y.C., et al., *Turbulent Flame Speed as an Indicator for Flashback Propensity of*
429 *Hydrogen-Rich Fuel Gases.* Journal of Engineering for Gas Turbines and Power-
430 Transactions of the Asme, 2013. **135**(11).
 - 431 7. Eichler, C., G. Baumgartner, and T. Sattelmayer, *Experimental Investigation of*
432 *Turbulent Boundary Layer Flashback Limits for Premixed Hydrogen-Air Flames*
433 *Confined in Ducts.* Journal of Engineering for Gas Turbines and Power-Transactions
434 of the Asme, 2012. **134**(1).
 - 435 8. Major, B., Powers, B., *Cost Analysis of NOx Control Alternatives for Stationary Gas*
436 *Turbines.* 1999.
 - 437 9. Cocchi, S., et al., *Experimental Characterization of a Hydrogen Fuelled Combustor*
438 *with Reduced No(X) Emissions for a 10 Mw Class Gas Turbine.* Proceedings of the
439 Asme Turbo Expo 2008, Vol 3, Pts a and B, 2008: p. 991-1000.
 - 440 10. S. Sigali, N.R., G. Sonato. *Hydrogen Combustion in Gas Turbines.* in *TOTeM34 - Gas*
441 *Turbine Research: Fuels, Combustion, Heat Transfer and Emissions.* 2010. Cardiff
442 University and CU Gas Turbine Research Centre, Wales: IFRF.
 - 443 11. Wu, J.F., et al., *Advanced gas turbine combustion system development for high*
444 *hydrogen fuels.* Proceedings of the Asme Turbo Expo, Vol 2, 2007: p. 1085-1091.
 - 445 12. Gazzani, M., et al., *Using Hydrogen as Gas Turbine Fuel: Premixed Versus Diffusive*
446 *Flame Combustors.* Journal of Engineering for Gas Turbines and Power-Transactions
447 of the Asme, 2014. **136**(5).
 - 448 13. Anantharaman, R.e.a., *European best practice guidelines for assessment of CO2*
449 *capture technologies.* 2011.

- 450 14. Therkelsen, P., et al., *Analysis of NO_x Formation in a Hydrogen-Fueled Gas Turbine*
451 *Engine*. Journal of Engineering for Gas Turbines and Power-Transactions of the
452 Asme, 2009. **131**(3).
- 453 15. Bozzelli, J.W. and A.M. Dean, *O+N₂H - a Possible New Route for Nox Formation in*
454 *Flames*. International Journal of Chemical Kinetics, 1995. **27**(11): p. 1097-1109.
- 455 16. Guo, H.S., et al., *The effect of hydrogen addition on flammability limit and NO_x*
456 *emission in ultra-lean counterflow CH₄/air premixed flames*. Proceedings of the
457 Combustion Institute, 2005. **30**: p. 303-311.
- 458 17. Li, H.L., M. Ditaranto, and D. Berstad, *Technologies for increasing CO₂*
459 *concentration in exhaust gas from natural gas-fired power production with post-*
460 *combustion, amine-based CO₂ capture*. Energy, 2011. **36**(2): p. 1124-1133.
- 461 18. Li, H.L., et al., *Impacts of exhaust gas recirculation (EGR) on the natural gas*
462 *combined cycle integrated with chemical absorption CO₂ capture technology*. 10th
463 International Conference on Greenhouse Gas Control Technologies, 2011. **4**: p. 1411-
464 1418.
- 465 19. Li, H.L., M. Ditaranto, and J.Y. Yan, *Carbon capture with low energy penalty:*
466 *Supplementary fired natural gas combined cycles*. Applied Energy, 2012. **97**: p. 164-
467 169.
- 468 20. Ditaranto, M., H. Li, and Y. Hu, *Evaluation of a Pre-combustion Capture Cycle Based*
469 *on Hydrogen Fired Gas Turbine with Exhaust Gas Recirculation (EGR)*. Energy
470 Procedia, 2014. **63**(0): p. 1972-1975.
- 471 21. ElKady, A.M., et al., *Application of Exhaust Gas Recirculation in a DLN F-Class*
472 *Combustion System for Postcombustion Carbon Capture*. Journal of Engineering for
473 Gas Turbines and Power-Transactions of the Asme, 2009. **131**(3).
- 474 22. Sundkvist, S.G., et al., *Concept for a Combustion System in Oxyfuel Gas Turbine*
475 *Combined Cycles*. Journal of Engineering for Gas Turbines and Power-Transactions of
476 the Asme, 2014. **136**(10).
- 477 23. Lefebvre, A.H. and D.R. Ballal, *Gas turbine combustion: alternative fuels and*
478 *emissions*. 2010, Boca Raton: Taylor & Francis. 557.
- 479 24. Ditaranto, M., J. Hals, and T. Bjorge, *Investigation on the in-flame NO reburning in*
480 *turbine exhaust gas*. Proceedings of the Combustion Institute, 2009. **32**: p. 2659-2666.
- 481 25. DigAnaRS: Delaware, U., 2013, *DARS - Software for Digital Analysis of Reactive*
482 *Systems*. 2013.
- 483 26. GRIMech.; Available from: http://www.me.berkeley.edu/gri_mech/.
- 484 27. Rortveit, G.J., et al., *Effects of diluents on NO_x formation in hydrogen counterflow*
485 *flames*. Combustion and Flame, 2002. **130**(1-2): p. 48-61.
- 486 28. Karbasi, M. and I. Wierzba, *The effects of hydrogen addition on the stability limits of*
487 *methane jet diffusion flames*. International Journal of Hydrogen Energy, 1998. **23**(2):
488 p. 123-129.
- 489 29. Ren, J.Y., et al., *Strain-rate effects on hydrogen-enhanced lean premixed combustion*.
490 Combustion and Flame, 2001. **124**(4): p. 717-720.
- 491 30. Schefer, R.W., *Hydrogen enrichment for improved lean flame stability*. International
492 Journal of Hydrogen Energy, 2003. **28**(10): p. 1131-1141.
- 493 31. York, W.D., W.S. Ziminsky, and E. Yilmaz, *Development and Testing of a Low NO_x*
494 *Hydrogen Combustion System for Heavy-Duty Gas Turbines*. Journal of Engineering
495 for Gas Turbines and Power-Transactions of the Asme, 2013. **135**(2).
- 496
- 497