

1 STATE OF RESEARCH ON SHEAR STRENGTHENING OF RC BEAMS WITH FRCM  
2 COMPOSITES

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14  
15 **ABSTRACT**

16 This paper summarizes the state of research on the topic of shear strengthening of RC beams using externally  
17 bonded FRCM composites. In the first part of this paper, a detailed bibliographical review of the literature on the  
18 shear strengthening of RC beams using FRCM composites is carried out, and a database of experimental tests is  
19 developed. Analysis of the database shows that FRCM composites are able to increase the shear strength of RC  
20 beams. The effectiveness of the strengthening system appears to be influenced by parameters including the  
21 wrapping configuration, matrix compressive strength relative to the concrete compressive strength, and axial  
22 rigidity of the fibers. Different failure modes have been reported, including fracture of the fibers, detachment of  
23 the FRCM jacket (with or without concrete attached), and slippage of the fibers through the mortar. A possible  
24 interaction between the internal transverse steel reinforcement and the FRCM system has also been observed. In  
25 the second part of this paper, four design models proposed to predict the contribution of the FRCM composite to  
26 the shear strength of RC beams are assessed using the database developed. Results show that the use of the  
27 properties of the FRCM composite in Models 3 and 4 instead of the fiber mechanical characteristics does not  
28 significantly increase the accuracy of the models. A simple formulation such as that proposed by Model 1, based  
29 on the bare fiber properties, is found to be more accurate for beams with or without composite detachment.

30 **KEYWORDS**

31 FRCM, FRP, reinforced concrete, shear, strengthening.

## 32 1. INTRODUCTION

33 Reinforced concrete (RC) structures are affected by external factors such as lack of maintenance, environmental  
34 conditions, or overloading that can cause deterioration and potentially diminish their structural performance. In  
35 addition, there is a growing need for upgrading existing structures in order to comply with requirements  
36 established in new design guidelines or to achieve an adequate level of performance due to the modification of  
37 expected loads caused by a change in use. The intervention of these structures requires the use of satisfactory  
38 rehabilitation and/or strengthening techniques that result in adequate behavior of the structure after the retrofitting  
39 process is carried out. Traditional techniques such as the increase of concrete section using concrete jackets or the  
40 use of externally bonded steel elements, which are common especially in developing countries, can often be  
41 considered as structurally acceptable but may not comply with modern requirements in which time- and cost-  
42 efficient interventions are usually required.

43 For this reason, externally-bonded fiber reinforced polymer (FRP) composites have become one of the most  
44 common intervention techniques for RC structures. Advantages of this technique include high stiffness-to-weight  
45 and strength-to-weight ratios, good fatigue characteristics, and ease of application. However, some limitations of  
46 this method, mainly related to the use of organic resins, have been pointed out [1]: (1) debonding of FRP from the  
47 concrete substrate; (2) poor behavior of epoxy resins at temperatures at or above the glass transition temperature;  
48 (3) relatively high cost of epoxy resins; (4) difficulty to apply onto wet surfaces or at low temperatures; (5) lack  
49 of vapor permeability; (6) incompatibility of epoxy resins with the substrate material; and (7) difficulty to conduct  
50 post-earthquake assessment of damage suffered by the structure. This suggests that the use of FRP might not be  
51 suitable for all applications, and new techniques that overcome some of these limitations are needed.

52 Composite materials that employ an inorganic cement-based matrix instead of an organic matrix allow for  
53 overcoming some of the limitations of FRP composites. Different names have been used in the literature to  
54 describe this type of composite depending on the matrix and fibers employed including textile reinforced concrete  
55 (TRC), textile reinforced mortar (TRM), fiber reinforced concrete (FRC), mineral based composites (MBC), and  
56 fiber reinforced cementitious matrix (FRCM). In this paper, the term FRCM is used to describe the aforementioned  
57 systems. FRCM composites exhibit significant heat resistance and vapor permeability and can be applied at low  
58 temperatures or onto wet surfaces [2]. The use of FRCM composites as a strengthening material for RC beams  
59 was first studied by [3-6], and their work can be considered as the starting point for the development of more  
60 recent research since their findings showed promising results. While research on the topic is still scarce, recent  
61 studies by [7-10], among others, have confirmed the effectiveness of this technique for flexural and shear

62 strengthening and confinement of axially/eccentrically loaded RC elements.  
 63 This paper summarizes the state of research on the topic of shear strengthening of RC beams using externally  
 64 bonded FRCM composites with the goal of serving as a reference point for the development of future research. In  
 65 the first part of this paper, a detailed bibliographical review of the literature on the shear strengthening of RC  
 66 beams using FRCM composites is carried out. This review summarizes the major findings and points out main  
 67 aspects that should be addressed in future research. In the second part of this paper, design models proposed to  
 68 predict the contribution of the FRCM composite to the shear strength of RC beams, including the ACI 549.4R  
 69 [11] expressions, are assessed using a database of experimental results collected and compiled by the authors.

## 70 2. EXPERIMENTAL DATABASE

71 Fifteen articles related to shear strengthening of RC beams using FRCM composites were found in the technical  
 72 literature and are summarized in Table 1. From these articles, a database that includes the characteristics and  
 73 results of experimental tests of the FRCM strengthened beams was developed and is presented in Appendix A.  
 74 Eighty-nine strengthened beams are included in the database.

75 Table 1 Summary of studies on shear strengthening of RC beams using externally bonded FRCM composites

Reference	Year	Beam Cross-sectional Shape <sup>a</sup>	Number of Strengthened Beams	Failure Mode		Strengthening Configuration <sup>d</sup>		
				Flexure <sup>b</sup>	Shear <sup>c</sup>	SB	U <sup>(e)</sup>	W
[9]	2006	R	3	2	1			3
[10]	2006	R	2		2			2
[12]	2008	T	9		9		9	(6)
[13]	2009	R	7		7	7		
[1]	2012	R	8		8	8		
[14]	2013	R	6	2	4	2	4	
[15]	2014	R	6		6	3	3	
[16]	2014	R	2		2		2	(1)
[17]	2014	T	10		10		10	(6)
[18]	2015	R	6		6		6	
[19]	2015	R	8		8	2	6	
[20]	2015	R	7	2	5		7	
[21]	2015	R	8	1	7	3	3	2
[22]	2015	R	1		1			1
[23]	2016	R	6		6	6		1
<b>Total</b>			89	7	82	31	50	(13)

<sup>a</sup>R=Rectangular, T-beam

<sup>b</sup>Yielding of longitudinal reinforcing steel bars followed by concrete crushing

<sup>c</sup>Failure mode related to FRCM debonding, fiber rupture, diagonal tension, and/or yielding of internal stirrups.

<sup>d</sup>SB=Side bonded, U=U-wrapped, W= Fully wrapped

<sup>e</sup>Numbers in parentheses indicate tests that include anchors

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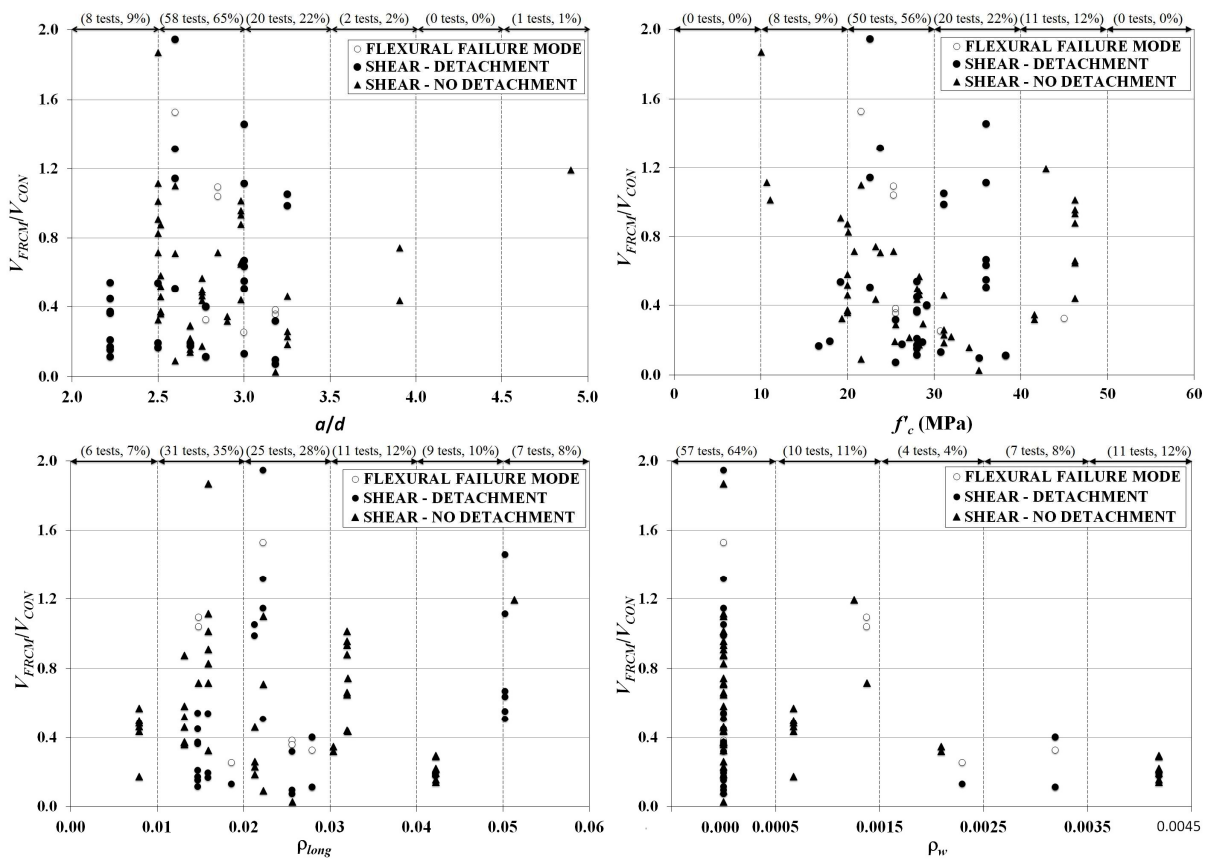
## 77 2.1 Evaluation of the Database and Distribution of Data

78 In order to evaluate the information collected in the database, the shear strength provided by the FRCM system  
79 ( $V_{FRCM}$ ) is calculated by subtracting the shear strength of the corresponding control beam ( $V_{CON}$ ) for each test.  
80 Although experimental specimens aimed to investigate the shear behavior of strengthened specimens are designed  
81 to attain shear failure, it is important to highlight that in some cases (seven tests, see Tables 1 and A1) the addition  
82 of the FRCM system changed the mode of failure from a brittle shear failure to a more ductile flexural failure.  
83 Specimens that failed in flexure can be considered as a lower bound of the strengthening capacity, but the behavior  
84 of beams that failed in that fashion is not further discussed in this paper.

85 Figures 1 to 3 present the variation of the ratio  $V_{FRCM}/V_{CON}$  as a function of the main geometrical and mechanical  
86 properties of the strengthened beams and the FRCM system. The horizontal axis of each plot is subdivided in  
87 order to evaluate the number and percentage of tests in different ranges, and values of which are labelled along  
88 the top of each graph. The points are subdivided according to the type of failure: flexural or shear. Shear failure  
89 is divided according to the presence or absence of detachment of the FRCM system from the strengthened beam.  
90 Further discussion on failure modes is presented in Section 2.2. It is important to note that the observations  
91 presented herein are based on the number and distribution of tests collected in the database and need to be validated  
92 when more test results become available

93 Figure 1 presents the variation of  $V_{FRCM}/V_{CON}$  as a function of the geometrical and mechanical properties of the  
94 beams ( $a/d$ =shear span to effective depth ratio;  $f'_c$ =mean cylinder compressive strength of concrete;  
95  $\rho_{long}$ =longitudinal steel reinforcement ratio,  $A_s/b_w d$ ; and  $\rho_w$ =internal transverse steel reinforcement ratio,  $A_w/b_w s$ ,  
96 where  $A_s$ =longitudinal steel reinforcement area;  $b_w$ =beam width;  $A_w$ =internal transverse steel reinforcement area;  
97 and  $s$ =internal transverse steel reinforcement spacing). Figure 1 shows that the increase in shear strength attributed  
98 to the FRCM system varies from 3% to 195% with an average value of 55%. While  $a/d$  varies from 2.22 to 4.90,  
99 most specimens (65%) have values of  $a/d$  between 2.5-3.0, and 87% between 2.5-3.5, which is common for the  
100 evaluation of shear strength of RC beams. In addition, as shown by Kani [24], the transition point between beam  
101 action and arch action corresponds to  $a/d$  values ranging from 2.5 to 3.0, which also corresponds to the lowest  
102 values of shear strength in terms of average shear stress. Therefore, beams with values of  $a/d$  in this range are  
103 usually used in research to obtain a lower bound of the shear strength. For the range of  $a/d$  tested, no clear relation  
104 can be observed between  $V_{FRCM}/V_{CON}$  and  $a/d$ . 56% of the tests were performed on beams with  $f'_c$  ranging from  
105 20-30 MPa and 78% from 20-40 MPa. These values of  $f'_c$ , which are relatively low for new structures, can be  
106 considered adequate to represent compressive strengths of many existing structures. 58% of the tests were

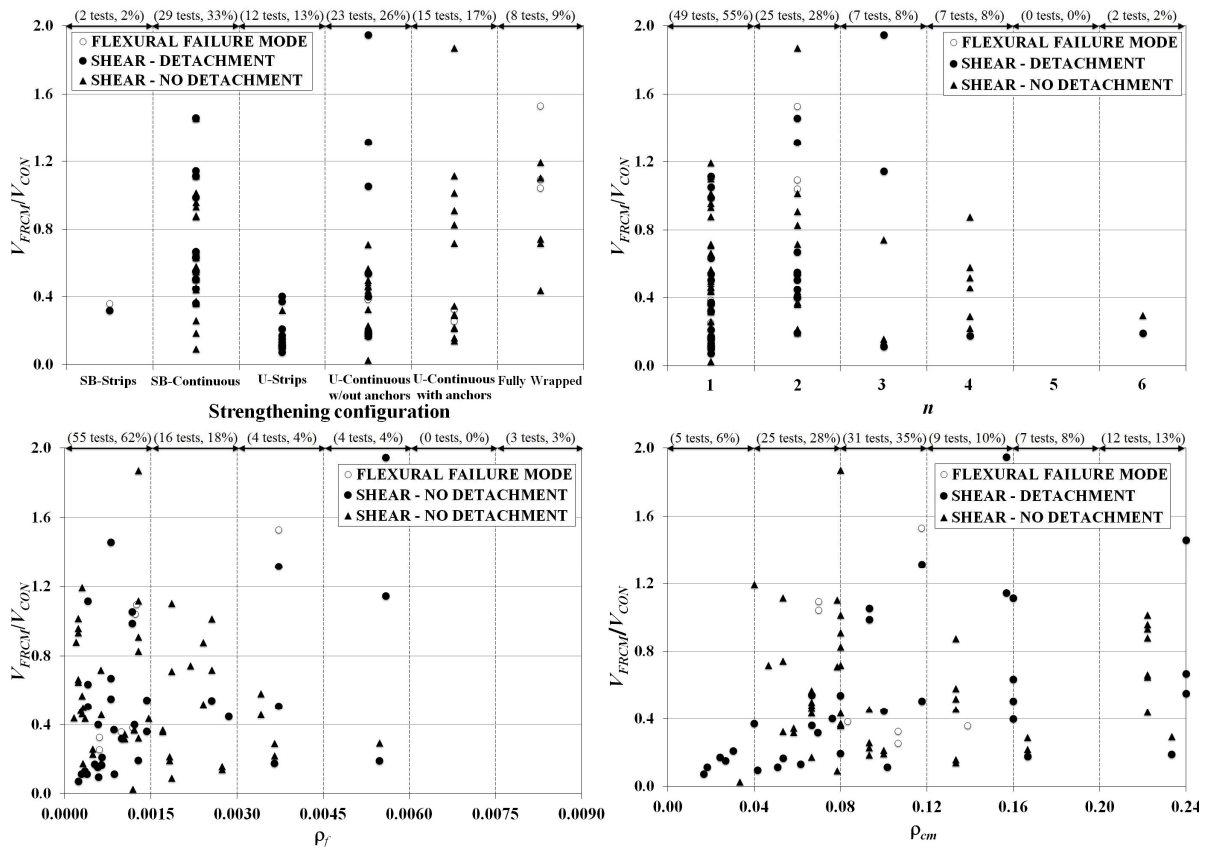
107 performed on beams that had a relatively high reinforcement ratio ( $\rho_{long} > 0.02$ ). Although beams with such large  
 108 values of  $\rho_{long}$  are not desirable in real applications, their use is explained by the experimental objective of avoiding  
 109 failure by bending. Only 36% of the tests were performed on beams with transversal steel reinforcement ( $\rho_w \neq$   
 110 0.0). Unlike the previous variables, and disregarding the beams with  $\rho_w = 0.0$ , a possible relationship between  $\rho_w$   
 111 and  $V_{FRCM}/V_{CON}$  can be observed. It appears that presence of a more dense distribution of stirrups ( $\rho_w > 0.0015$ )  
 112 reduces the effectiveness of the FRCM system. An explanation for this behavior is the possible interaction between  
 113 the internal transverse steel reinforcement and the external FRCM strengthening, which has been reported for FRP  
 114 composites [25-27]. A more detailed description of this phenomenon is presented in Section 2.3.



115  
 116 Figure 1 Variation of  $V_{FRCM}/V_{CON}$  with  $a/d$ ,  $f'_c$ ,  $\rho_{long}$ , and  $\rho_w$   
 117

118 Figure 2 presents the variation of  $V_{FRCM}/V_{CON}$  as function of the geometrical properties of the strengthening system  
 119 (strengthening configuration;  $n$ =number of fiber layers;  $\rho_f$ =fiber reinforcement ratio,  $2nt_f w_f/b_w s_f$ ; and  $\rho_{cm}$ =FRCM  
 120 reinforcement ratio,  $2t_{cm} w_f/b_w s_f$ , where  $t_f$ =nominal thickness of fiber sheets;  $w_f$ =width of FRCM strips;  $s_f$ =spacing  
 121 of FRCM strips; and  $t_{cm}$ = total thickness of the FRCM composite  $(n+1)*t_m$  with  $t_m$  the nominal thickness of a  
 122 matrix layer). Most tests have been performed on beams strengthened with continuous side bonded (33%)  
 123 configurations or continuous U-jackets with (26%) or without anchors (17%) configurations. Comparing the

124 additional shear strength  $V_{FRCM}$  relative to  $V_{CON}$  for these two configurations, although slightly higher values of  
 125  $V_{FRCM}/V_{CON}$  are related to the U-continuous configuration, it is not possible to conclude that using this  
 126 configuration will guarantee a better performance of the strengthened beam, which agrees with [15] who  
 127 concluded that side bonded and U-wrapped configurations showed similar performance in terms of strength. In  
 128 side bonded configurations detachment of the FRCM composite was less frequently observed, while for U-  
 129 wrapped configuration most failures were accompanied by composite detachment, either at the composite-  
 130 substrate interface or within the substrate. Although the experimental evidence is more limited, a similar behavior  
 131 is also observed in beams strengthened with strips. The use of anchors with the U-wrapped configuration appears  
 132 to mitigate detachment of the composite. A more detailed analysis regarding the type of failure mode and the  
 133 influence of anchors is discussed in Section 2.2.



134

135

136

Figure 2 Variation of  $V_{FRCM}/V_{CON}$  with strengthening configuration, number of layers  $n$ ,  $\rho_f$ , and  $\rho_{cm}$

137 55% of the tests were carried out on beams strengthened with one layer of FRCM composite. Although some  
 138 higher values of  $V_{FRCM}/V_{CON}$  can be seen increasing the number of layers from 1 to 2 or 3, the effectiveness of the  
 139 system appears to be reduced when a larger number of layers are provided, i.e., the gain in shear strength may not  
 140 be proportional to the number of layers. An increasing trend is observed with the increase of  $\rho_{cm}$  (that can be  
 141 understood as a relative increase in the width of the concrete cross section) implying that the increase in

142  $V_{FRCM}/V_{CON}$  depends not only on the amount of fibers included but also on the thickness of the cementitious matrix  
 143 applied. Thicker layers of cementitious matrix, i.e. larger values of  $\rho_{cm}$ , imply a larger increase in the concrete  
 144 section, and therefore an increase in the capacity of the beam would be expected, even if no fibers were included  
 145 [13]. This trend is clearer for beams that failed by detachment of the composite.

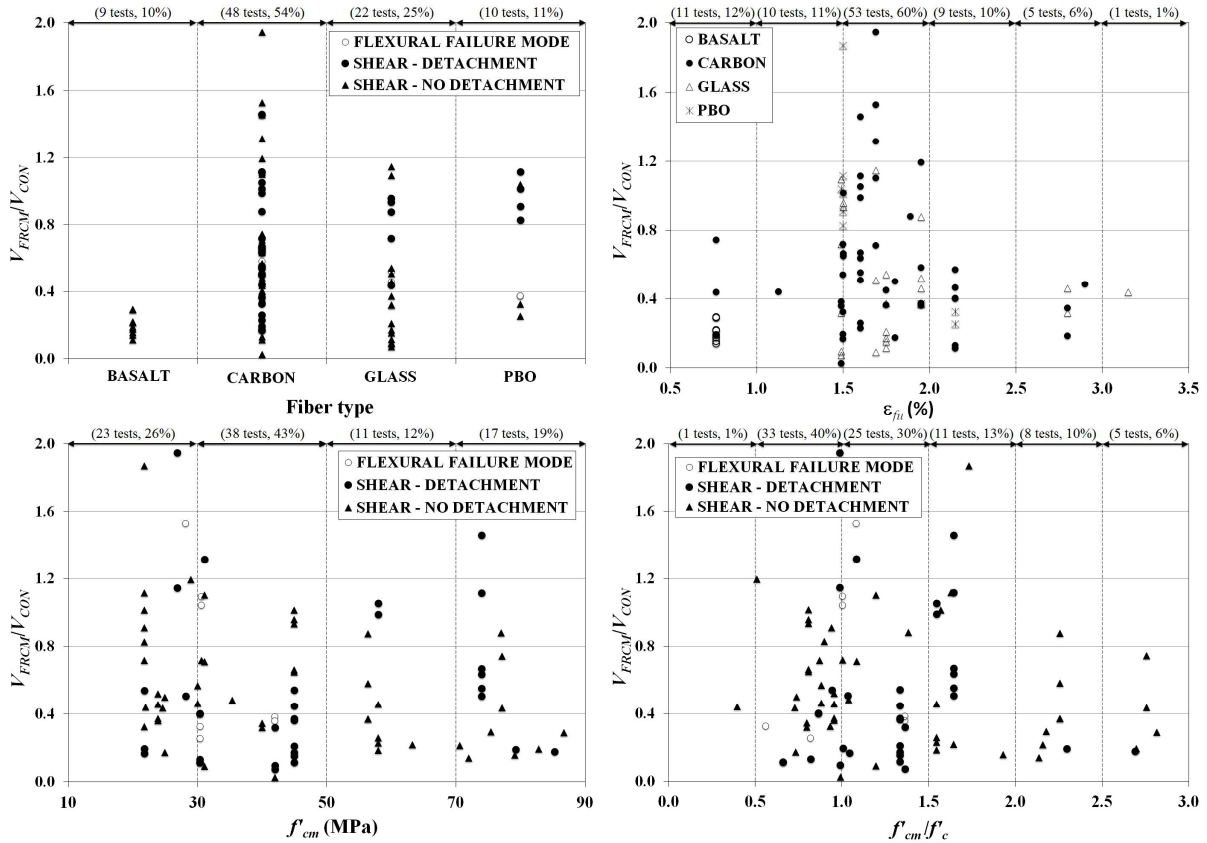


Figure 3 Variation of  $V_{FRCM}/V_{CON}$  with fiber type, fiber ultimate strain  $\epsilon_{fi}$ ,  $f'_{cm}$ , and  $f'_{cm}/f'_c$ .

146  
 147  
 148

149 In Figure 3, the influence of key mechanical properties of the FRCM composite (fiber type; bare fiber ultimate  
 150 strain ( $\epsilon_{fi}$ ); cementitious matrix compressive strength ( $f'_{cm}$ ); and ratio  $f'_{cm}/f'_c$ ) on the ratio  $V_{FRCM}/V_{CON}$  is presented.  
 151 Tests on beams with carbon fiber represent 48% of the available data, followed by glass, PBO, and basalt fibers.  
 152 An important observation regarding fiber type is that beams strengthened with carbon FRCM were capable of  
 153 achieving larger increases in shear strengths than beams strengthened with other type of fibers. Regarding the  
 154 values of  $\epsilon_{fi}$ , it is interesting to notice that even though there is a large dispersion for all the fibers, most tests are  
 155 concentrated in a range from 1.5 to 2.0% (60% of the tests). In addition, for carbon, glass, and PBO fibers, the  
 156 larger values of  $V_{FRCM}/V_{CON}$  ratios are also concentrated in the same range. Large values of  $f'_{cm}$  appear to be related  
 157 to a lower effectiveness of the system (i.e., lower  $V_{FRCM}/V_{CON}$  values). A similar trend is observed when the ratio  
 158  $f'_{cm}/f'_c$  is analyzed. It can be seen that better performance might be achieved when the compressive strengths of  
 159 the substrate and the cementitious matrix are similar ( $f'_{cm}/f'_c$  close to 1.0).

## 160 2.2 Failure Modes of FRCM Strengthened Beams

161 Regarding fully wrapped beams, [9] reported fiber rupture and observed beam cracking clearly visible on the  
162 surface of the FRCM jacket. These findings are corroborated by [10] who reported a similar behavior. This type  
163 of failure agrees with the experimental evidence for beams fully wrapped with FRP composites, which tend to fail  
164 due to FRP rupture [27]. As noted by [28] for FRP composites, this behavior indicates that the wrapping  
165 configuration is able to provide significant anchorage to avoid composite debonding. It is worth mentioning that  
166 information on the overlap length and its design is not reported in the references but should be related to the  
167 effective length of the composite, i.e., the length needed to fully develop the load-carrying capacity of the interface  
168 [29].

169 It is not as straightforward to identify a typical failure mode for side bonded and U-wrap configurations as it is  
170 for fully wrapped beams. Composite detachment, which is described as debonding of the FRCM jacket from the  
171 substrate (with or without concrete attached) in this paper, is reported in some of the references [14,19,20]. In  
172 most cases, detachment was located at the matrix-substrate surface without affecting the concrete surface,  
173 although peeling off of the concrete cover (i.e., within the substrate) has also been witnessed [21,23]. However, it  
174 is not possible to conclude that failure will be exclusively related to this phenomenon as other failure modes have  
175 also been reported in the available literature. [1,16,18] described failure caused by diagonal tension. The same  
176 behavior, together with rupture of some fibers, was observed by [13]. Azam and Soudki [15] described failure by  
177 diagonal tension associated with a large diagonal crack for most of their specimens, although the two beams that  
178 reached a higher shear strength experienced detachment and shear compression failure. Tetta *et al.* [21] reported  
179 slippage of the vertical fibers through the mortar and partial fiber rupture. According to their findings, the type of  
180 failure depends on the strengthening configuration with slippage being more pronounced in side bonded  
181 configurations and almost eliminated for fully wrapped configurations in which fiber rupture is the dominating  
182 failure mechanism. Fiber slippage is another form of debonding that has been observed in some types of FRCM  
183 composite-concrete joints [29,30,31].

184 Table 2 summarizes the type of failure mode reported for the different strengthening configurations for beams  
185 without anchors. It is interesting to note that the failure mode reported for most of the side bonded configurations  
186 was not related to the detachment of the FRCM composite from the substrate. This behavior does not agree with  
187 the findings for beams strengthened with FRP composites where two- or three-sided jackets fail mainly by  
188 debonding of the composite [28]. In fact, some codes for the design of externally bonded FRP composites do not  
189 allow the use of side bonded configurations for shear strengthening of RC beams [32] in order to avoid an early



190 debonding of the system. For U-wrapped strengthened beams the prevailing failure mode is associated with  
 191 detachment of the composite, although failure without detachment was reported in 35% of the tests. Considering  
 192 that all unstrengthened control beams failed in shear, it is also interesting to note that the ability to transform this  
 193 type of failure into a flexural failure is not exclusively limited to fully wrapped beams, although it has been rarely  
 194 reported in side bonded beams.

195 Table 2 Failure modes of beams with different FRCM composite strengthening configurations

Strengthening Configuration	Failure Mode		
	Detachment	No Detachment	Flexure
Side bonded	12	18	1
U-wrapped <sup>a</sup>	21	13	3
Fully wrapped	0	5	3

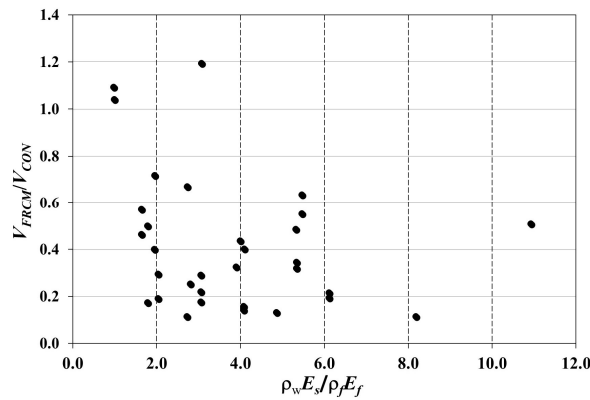
196 <sup>a</sup>Beams with anchors are not included  
 197

### 198 2.3 Interaction Between Internal and External Shear Reinforcement

199 It has been noted that the interaction between internal steel reinforcement and external FRP reinforcement should  
 200 be taken into account to properly predict the overall shear strength of a strengthened beam [33,34]. According to  
 201 [35], the maximum contributions of steel stirrups and FRP to the shear strength are not reached simultaneously  
 202 implying that their combined contribution may actually be less than the simple summation of their respective  
 203 values. The possible interaction between the internal and external shear reinforcement for FRCM systems has also  
 204 been reported by [13,20], who witnessed a significant reduction in the strain values measured in the stirrups of the  
 205 strengthened beams when compared with the control beams at the same load levels. In fact, for the beams tested  
 206 by [20], the presence of the FRCM system precluded yielding of the stirrups, as has also been reported for similar  
 207 beams strengthened with FRP composites [25].

208 The ratio of the axial stiffness of the transverse steel reinforcement to that of the FRP composite ( $\rho_w E_s / \rho_f E_f$  where  
 209  $E_s$ =elastic modulus of stirrups, and  $E_f$ =elastic modulus of the bare fibers) has been used to evaluate the internal  
 210 and external shear reinforcement interaction in FRP-strengthened beams. For FRP composites the effectiveness  
 211 of the strengthening system reduces when the ratio  $\rho_w E_s / \rho_f E_f$  increases [25]. The same trend is observed for FRCM  
 212 composites in Figure 4, in which  $V_{FRCM} / V_{CON}$  is plotted against  $\rho_w E_s / \rho_f E_f$  for strengthened beams with stirrups.  
 213 Results in Figure 4 suggest that, for a given amount of FRCM, increasing the amount of internal reinforcement  
 214 decreases the contribution of the FRCM (i.e., lower  $V_{FRCM} / V_{CON}$  values). Having a larger internal transversal steel  
 215 reinforcement ratio by providing a smaller stirrup spacing implies that more stirrups will be crossed by the critical  
 216 shear crack, and they might not yield before failure of the beam. In other words, the internal shear reinforcement  
 217 may not be able to achieve its design value (based on the assumption of yielding) and provide the same

218 contribution it gives in the unstrengthened element. This implies that subtracting the control beam shear strength  
 219 from the total shear strength of the strengthened beams in order to obtain  $V_{FRCM}$  may not accurately reflect the  
 220 contribution of the FRCM system.



221

222 Figure 4 Variation of  $V_{FRCM}/V_{CON}$  with  $\rho_w E_w / \rho_f E_f$  for strengthened beams with stirrups  
 223

224 **2.4 Anchorage Systems**

225 The few studies that have included anchors for the FRCM composite shear strengthening system have shown  
 226 mixed results. Baggio *et al.* [16] evaluated the efficiency of FRP spike anchors for rectangular beams strengthened  
 227 in shear with U-wrapped FRCM composites. The anchors, composed of carbon fibers, were inserted in predrilled  
 228 holes and then fanned out. The beam with anchors showed an increase of only 3% over the strengthened beam  
 229 without anchors. Although beams with and without anchors exhibited a diagonal tension shear failure, the presence  
 230 of the anchors slightly changed the inclination of the shear crack around the anchors. Considering that failure of  
 231 the strengthened beams by fiber slippage has been reported for certain FRCM composites [29,30,31], the lack of  
 232 effectiveness of this type of anchor may be linked to the fact that they are intended to restrain out-of-plane peeling  
 233 of the composite and do not restrain the in-plane fiber slippage [36].

234 L-shaped steel sections were used by [12] to anchor the FRCM system for U-wrapped T-beams. One leg of the  
 235 steel section was glued to the FRCM composite, while the other was anchored to the bottom of the beam flange  
 236 by means of vertical steel bars installed in pre-drilled holes through the entire thickness of the flange. For beams  
 237 without anchors, the increase in shear capacity of the beam was approximately 19%, independent of the number  
 238 of fiber layers. For beams with anchors, the shear increase strength ranged between 14% and 29%, depending on  
 239 the number of layers. Although higher strengths were achieved for certain beams with anchors, the results were  
 240 not consistent. However, the presence of the anchors reportedly avoided the FRCM system detachment.

241 Tzoura and Triantafillou [17] used a 3 mm thick curved steel section fixed to the slab with threaded rods to anchor  
 242 FRCM U-wrapped T-beams. The steel sections were placed at the corners between the slab and the beam web.

243 The rods were placed inside 45° holes filled with an epoxy adhesive at a fixed spacing. A significant increase in  
244 the effectiveness of the FRCM jackets for the beams with anchors was reported. For beams strengthened with low  
245 textile density, the increase in strength appeared to be more significant, from approximately 18% for beams  
246 without anchors to a maximum of 187% when anchors were present. For beams with high textile density, the  
247 increase in shear strength ranged from 32% for beams without anchors to a maximum of 112% for specimens with  
248 anchors.

249

### 250 3. ASSESSMENT OF AVAILABLE MODELS

#### 251 3.1 Overview

252 Four models proposed to determine the contribution of the FRCM composite to the shear strength of RC beams  
253 are evaluated in this section: Model 1 by Triantafyllou and Papanicolaou [9], Model 2 by Escrig *et al.* [18], Model  
254 3 by ACI 549.4R [11], and Model 4 by Ombres [20]. Models 1 and 2 are based on the properties of the FRCM  
255 composite fibers, and Models 3 and 4 are based on properties of the composite, as discussed in Sections 3.2 and  
256 3.3, respectively. For the case of Model 3, which is the only guide available at this time for the design and  
257 construction of FRCM composites, the contribution to the shear strength provided by the strengthening system  
258  $V_{FRCM}$  is considered to be additive to the strength of the unstrengthened (control) beam ( $V_{CON}=V_c+V_s$ ), as shown  
259 in Eq. (1), in order to determine the total shear capacity of the strengthened beam  $V_n$ :

$$V_n = V_c + V_s + V_{FRCM} \quad (1)$$

261 where  $V_c$  and  $V_s$  are the contributions to the shear strength provided by the concrete and internal transversal steel  
262 reinforcement, respectively.

263 Currently there are no European standards for the evaluation of  $V_{FRCM}$ . However, it is worth noting that for the  
264 case of FRP strengthened beams,  $V_n$  in certain European-based approaches [37, 38] is computed including only  
265  $V_s$  and the contribution of the FRP system,  $V_f$ , and its value is limited by the shear strength of the concrete  
266 compression strut,  $V_{c,max}$  [39] as shown in Eq. (2).

$$V_n = \min\{V_s + V_f, V_{c,max}\} \quad (2)$$

267 Values of  $V_c$ ,  $V_s$ , and  $V_{c,max}$  in Eqs. (1) and (2) are calculated using the equations in current design provisions for  
268 unstrengthened RC beams. In this paper, the evaluation of the models is carried out considering only the strength  
269 provided by the FRCM system (i.e.  $V_{FRCM}$ ) and not the total shear capacity (i.e.  $V_n$ ) achieved after strengthening.  
270 Although the four models present different formulations, they are each based on the well-known truss analogy  
271 and differ mainly in the expression used to evaluate the stress (or strain) in the FRCM system along the critical

272 shear crack. Models 1 and 3 are based on a fixed angle of the diagonal shear crack relative to the longitudinal axis  
 273 ( $\theta$ ). Models 2 and 4 allow the use of variable angles, however only Model 2 was developed using angles different  
 274 from  $45^\circ$  when this information was provided in the articles used to calibrate the model; otherwise the value of  
 275  $\theta$  was set to  $45^\circ$  [18]. Therefore, and considering the limited data available reporting the actual value of  $\theta$ , a fixed  
 276 value of  $\theta=45^\circ$  is used in this paper to evaluate and compare the different models. It is also worth noting that in  
 277 practical design applications,  $\theta$  is unknown, and a fixed value of  $45^\circ$  is usually used.  
 278 For each model, average (AVG) values of test-to-predicted ratios of the term  $V_{FRCM}$ , denoted as  $V_{test}/V_{pred}$ , are  
 279 reported, as well as the standard deviation (STD) and coefficient of variation ( $COV_1$ ) computed with respect to a  
 280 mean value of 1, which implies a perfect match between  $V_{test}$  and  $V_{pred}$ , as shown in Eq. (3):

$$COV_1 = \sqrt{\frac{\sum_1^N \left( \frac{V_{test,i}}{V_{pred,i}} - 1 \right)^2}{N}} \quad (3)$$

281 where  $N$  is the number of tests. As per Section 2.1, the value  $V_{test}$  is calculated by subtracting the shear strength of  
 282 the corresponding control beam ( $V_{CON}$ ) for each test, whereas the value  $V_{pred}$  is computed by the model. In the  
 283 assessment of the models, strengthened beams that included anchors and/or those that failed in flexure were not  
 284 considered.

285 Different subsets of the complete database needed to be used in the assessment of the different models due to the  
 286 limitations of each model and the parameters included. As mentioned earlier in this section, Models 1 and 2 use  
 287 the properties of the bare fibers, and all references included in Tables 1 and A1 reported the required properties.  
 288 The assessment of Model 1 is made using all tests, except those with anchors or that failed in flexure, and the  
 289 resulting database is referred to as Database 1 (“DB1”), which includes 69 tests. Model 2, on the other hand, was  
 290 formulated based on tests in which detachment of the FRCM system from the substrate was prevented. For this  
 291 reason, its evaluation is carried out using a subset of DB1, referred to as Database 2, (“DB2”), which includes  
 292 only those tests that did not exhibit composite detachment (36 tests). The performance of Models 1 and 2 is then  
 293 compared using DB2 (Section 3.4), since it is common to both.

294 Models 3 and 4 evaluate the additional shear strength provided by the FRCM system based on the mechanical  
 295 properties of the FRCM system as a composite and are presented in Section 3.3. However, only four of the  
 296 references [14,18,19,20] reported the required properties of the FRCM composite. Unfortunately, the tests  
 297 presented by [14] had to be disregarded because the value reported for the elastic modulus of the FRCM composite  
 298 ( $E_{FRCM}=2.72$  GPa) was approximately 50 times lower than values reported for this variable in the available  
 299 literature, which resulted in values of  $V_{pred}$  that were clearly anomalous with respect to the other tests. Thus, a

300 subset of DB1, referred to as Database 3 (“DB3”) that includes 19 available tests from references that reported  
 301 the mechanical properties of the composite was used to evaluate Models 3 and 4. Comparison of Models 1, 3, and  
 302 4 is carried out using DB3 (Section 3.4), since it is common to all three models. Model 2 is not included in this  
 303 comparison because most tests in DB3 failed due to composite detachment of the FRCM system.

304 In order to facilitate the analysis, the formulations of the models are presented in this paper with a uniform notation.

### 305 3.2 Models based on properties of fibers

#### 306 3.2.1 Model 1: Triantafillou and Papanicolaou [9]

307 Model 1 was first presented for fully wrapped rectangular beams and then extended for U-wrapped beams [17].

308 Assuming that the fiber is comprised of perpendicular rovings aligned perpendicular and parallel to the beam  
 309 longitudinal axis,  $V_{FRCM}$  is given by Eq. (4):

$$V_{FRCM} = \rho_f \sigma_{eff} b_w d_f \quad (4)$$

310 where  $\rho_f$  is the geometrical reinforcement ratio of composite material, and  $d_f$  is the effective depth of the jacket  
 311 taken as  $0.9d$  ( $d$ =effective depth) for rectangular beams or the height of the web for T-beams. The effective stress  
 312 in the FRCM system ( $\sigma_{eff}$ ) is computed based on the average strain reached across the shear crack. Based on  
 313 limited experimental evidence, [9] indicated that this strain is approximately 50% of the ultimate strain of the bare  
 314 fibers  $\varepsilon_{fu}$ , although they highlighted that further research is needed to validate this approximation. Therefore,  $\sigma_{eff}$   
 315 is computed by Eq. (5):

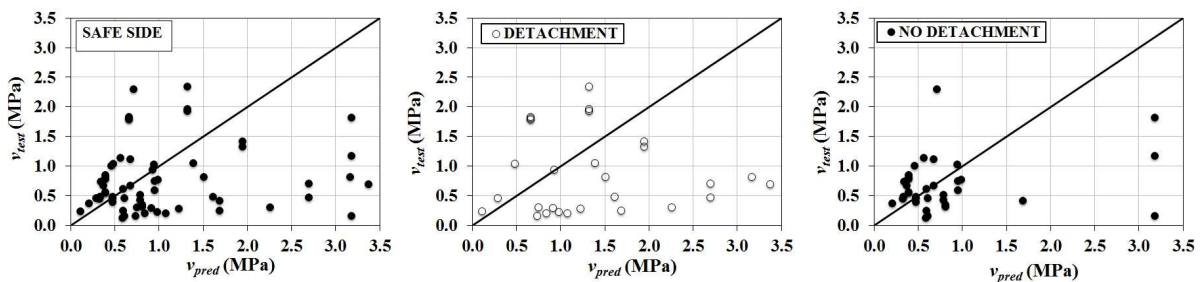
$$\sigma_{eff} = 0.5E_f \varepsilon_{fu} \quad (5)$$

316 Figure 5 compares the test versus predicted values provided by the FRCM system in term of average shear stress,  
 317  $v_{test}$  and  $v_{pred}$ , where  $v_{test}$  and  $v_{pred}$  are computed according to Eqs. (6a) and (6b), respectively. The solid line  
 318  $v_{test}/v_{pred}=1.0$  divides safe (points above the line) and unsafe (points below the line) values.

$$v_{test} = \frac{V_{test}}{b_w d} \quad (6a)$$

$$v_{pred} = \frac{V_{pred}}{b_w d} \quad (6b)$$

319



320

321 a) b) c)  
 322 Figure 5  $v_{test}$  versus  $v_{pred}$  for Model 1: a) DB1; b) DB1-Detachment; c) DB1-No detachment

323  
 324 For beams that failed by detachment of the strengthening system, Figure 5b shows that Model 1 tends to  
 325 overestimate (unsafe) the contribution of the FRCM composite, with AVG=0.80 (Table 3). This overestimation  
 326 indicates that actual strain in the fibers might be lower than 50% of the ultimate strain assumed by the model. For  
 327 beams with no detachment, the concentration of points around the solid line indicates a better agreement between  
 328 predicted and test values. The AVG value for beams with no detachment is 1.12, which indicates a slight  
 329 understimation (safe) of the FRCM composite contribution. Regarding the accuracy of the model, a larger value  
 330 of  $COV_1$  is associated with beams that failed by FRCM detachment.

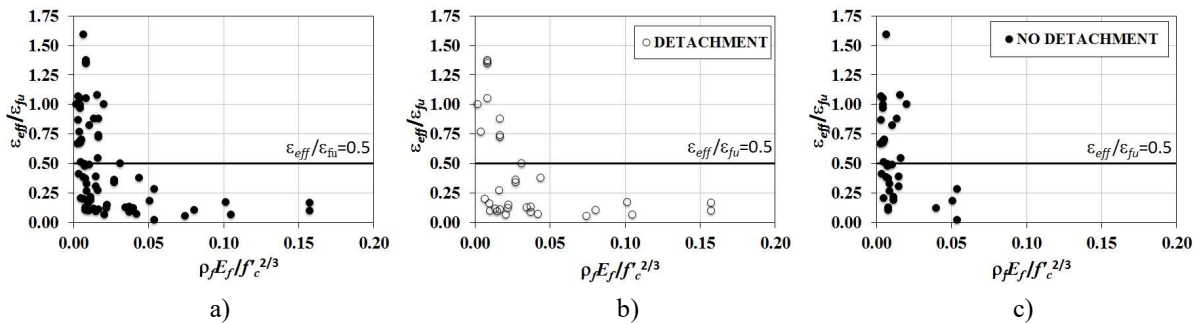
331 Table 3  $V_{test}/V_{pred}$  for Model 1 with DB1

Sample	#	AVG	STD	COV <sub>1</sub>
Detachment	33	0.80	0.75	0.86
No Detachment	36	1.12	0.71	0.72
Total	69	0.97	0.79	0.79

332  
 333 Considering the limited experimental evidence used by [9] to define the value of  $\sigma_{eff}$ , Eq. (7) is used to determine  
 334 the effective strain in the fibers  $\epsilon_{eff}$  for the tests included in DB1. Rearranging Eq. (4),  $\epsilon_{eff}$  can be calculated from  
 335 the value of  $V_{test}$  as:

$$\epsilon_{eff} = \frac{V_{test}}{\rho_f E_f b_w d_f} \quad (7)$$

336 The average value of  $\epsilon_{eff}$  normalized by  $\epsilon_{fu}$  (i.e.,  $\epsilon_{eff}/\epsilon_{fu}$ ), without including values of  $\epsilon_{eff}/\epsilon_{fu} > 1.0$ , is 0.38  
 337 (COV=0.86), which is lower than the factor 0.50 proposed by the model. However, as shown in Figure 5 and  
 338 Table 3, the failure mode of the beams influences the performance of the model.



340  
 341 Figure 6 Normalized fiber strain  $\epsilon_{eff}/\epsilon_{fu}$  versus  $\rho_f E_f / f_c^{2/3}$ : a) DB1; b) DB1-Detachment; c) DB1-No detachment  
 342 As expressed by [40] and adopted by the *fib* design model for FRP systems [37], the effective strain in the fibers  
 343 depends on the axial rigidity ( $E_f \rho_f$ ) and is inversely proportional to the tensile strength of the substrate expressed  
 344 as  $f_c^{2/3}$ . In Figure 6, the values of  $\epsilon_{eff}/\epsilon_{fu}$  are plotted in terms of the ratio  $\rho_f E_f / f_c^{2/3}$ , where  $\epsilon_{eff}$  is calculated using

346 Eq. (7). The constant value suggested by the model ( $\varepsilon_{eff}/\varepsilon_{fu}=0.5$ ) is also indicated in the graph. Figure 6a shows  
347 that the ratio  $\varepsilon_{eff}/\varepsilon_{fu}$  tends to decrease with increasing  $\rho_f E_f / f'_c{}^{2/3}$ , as has been found for FRP composites [40]. For  
348 beams that failed by detachment (Figure 6b),  $\varepsilon_{eff}$  is generally lower than 50% of  $\varepsilon_{fu}$ , with an average of 0.28  
349 (COV=0.85). For beams that did not show detachment, the average value is 0.46 (COV=0.58), which is close to  
350 the value proposed by the model, although the relationship of  $\varepsilon_{eff}/\varepsilon_{fu}$  and  $\rho_f E_f / f'_c{}^{2/3}$  is not as clear as for beams that  
351 failed by detachment. However, beams that did not show detachment generally present lower values of  $\rho_f E_f / f'_c{}^{2/3}$ .  
352 In fact, 80% of tests that did not fail by detachment present values of  $\rho_f E_f / f'_c{}^{2/3}$  lower than 0.02, while only 33%  
353 of beams with detachment fall in that range. For a constant concrete strength, this finding indicates that a less stiff  
354 strengthening solution, i.e. lower values of  $E_f \rho_f$ , might avoid the onset of detachment. Although both detachment  
355 and shear failure can be considered as brittle failures, a better exploitation of the system can be expected with  
356 larger values of effective strain, which are associated to beams with no detachment.

### 357 3.2.2 Model 2: Escrig et al. [18]

358 Model 2 computes  $V_{FRCM}$  according to Eq. (8):

$$359 \quad V_{FRCM} = 2n\varepsilon_{eff}E_f t_f d_f (\cot\alpha + \cot\theta) \sin^2\alpha \quad (8)$$

360 where  $\alpha$  is the fiber inclination angle with respect to the longitudinal axis of the beam, and the other variables  
361 were defined previously. Based on the research by [40] and using data collected from the literature for specimens  
362 without anchors that avoided composite detachment, [18] proposed the following equations for computing the  
363 effective strain in the fibers  $\varepsilon_{eff}$ :

- 364 • Fully wrapped:

$$\varepsilon_{eff} = 0.035 \left( \frac{f'_c{}^{2/3}}{E_f \rho_f} \right)^{0.65} \varepsilon_{fu} \quad (9)$$

- 365 • Side bonded or U-wrapped:

$$\varepsilon_{eff} = 0.020 \left( \frac{f'_c{}^{2/3}}{E_f \rho_f} \right)^{0.55} \varepsilon_{fu} \quad (10)$$

366 In Eqs. (9) and (10),  $E_f$  and  $f'_c$  are expressed in units of GPa and MPa, respectively. In Figure 7,  $v_{test}$  is plotted  
367 versus  $v_{pred}$  using Model 2 for the tests included in DB2, and Table 4 summarizes values of AVG, STD and COV<sub>1</sub>.

368

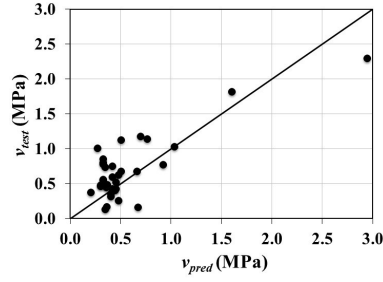


Figure 7  $v_{test}$  versus  $v_{pred}$  for Model 2 (DB2)

Table 4  $V_{test}/V_{pred}$  for Model 2 with DB2

Sample	#	AVG	STD	COV <sub>1</sub>
No Detachment	36	1.35	0.68	0.77

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373

374 For Model 2, the effective strain in the fibers can be computed from the value of  $V_{test}$  by rearranging Eq. (8) in the  
375 form of Eq. (11):

$$\varepsilon_{eff} = \frac{V_{test}}{2nE_f t_f d_f (\cot\alpha + \cot\theta) \sin^2\alpha} \quad (11)$$

376 In Figure 8, values of the ratio  $\varepsilon_{eff}/\varepsilon_{fu}$  are plotted against  $\rho_f E_f / f'_c{}^{2/3}$ , where  $\varepsilon_{eff}$  is calculated using Eq. (11), and are  
377 shown as “calculated” in the graph. Figure 8 also includes the normalized values of  $\varepsilon_{eff}$  computed using Eqs. (9)  
378 and (10) shown as “predicted” in the graph. The relationship between  $\varepsilon_{eff}/\varepsilon_{fu}$  and  $\rho_f E_f / f'_c{}^{2/3}$  is not clear for beams  
379 that did not show detachment. Although a possible increase of  $\varepsilon_{eff}/\varepsilon_{fu}$  with the decrease of  $\rho_f E_f / f'_c{}^{2/3}$  is observed,  
380 the points do not follow the trend depicted by Eqs. (9) and (10). It is worth mentioning that for a few specimens,  
381 the value of the ratio  $\varepsilon_{eff}/\varepsilon_{fu}$  is slightly larger than 1.0, implying that the effective strain is larger than the rupture  
382 strain. It should be noted that the value of  $\varepsilon_{eff}$  is not measured but rather determined by the model, and in some  
383 cases the value of the  $\varepsilon_{fu}$  is given by the manufacturer as a minimum value.

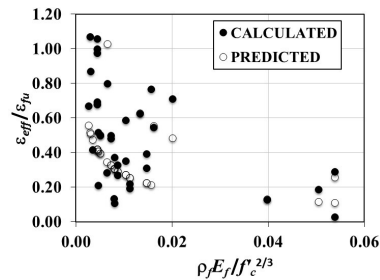


Figure 8 Normalized fiber strain  $\varepsilon_{eff}/\varepsilon_{fu}$  versus  $\rho_f E_f / f'_c{}^{2/3}$

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### 387 3.3 Models based on properties of the FRCM composite

#### 388 3.3.1 Model 3: ACI 549.4R [11]

389 The ACI 549.4R guideline [11] is currently the only guide for design and construction of FRCM systems. However,  
390 it is based on few experimental tests, and the guidelines note that the equations require further validation.



391 According to Model 3, the contribution to the shear strength of RC beams by continuous FRCM U-wrapped or  
 392 continuous fully wrapped composite is computed using Eq. (12):

$$V_{FRCM} = nA_f\sigma_{eff}d \quad (12)$$

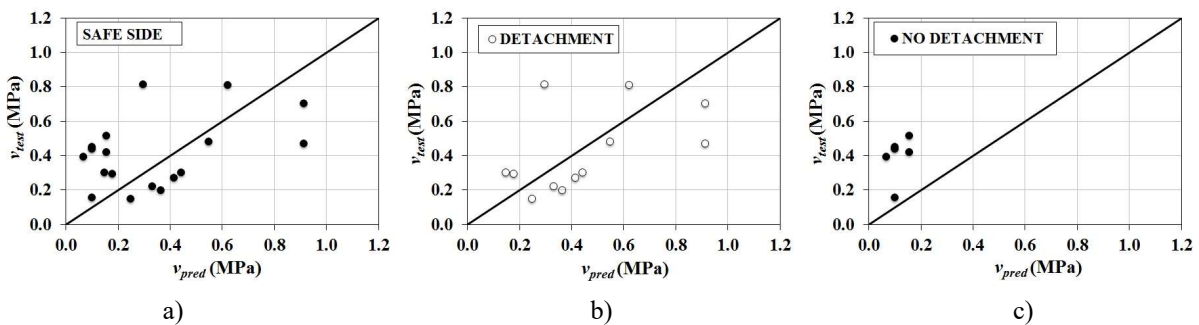
393 In Eq. (12),  $A_f$  is the area of mesh reinforcement per unit width effective in shear, and the other variables were  
 394 defined previously. The so-called design tensile strength of the FRCM shear reinforcement  $\sigma_{eff}$  depends on the so-  
 395 called design tensile strain of the reinforcement  $\varepsilon_{eff}$  and the tensile modulus of elasticity of the cracked FRCM  
 396 composite material  $E_{FRCM}$ , and is computed using Eqs. (13) and (14):

$$\varepsilon_{eff} = \varepsilon_{FRCM,u} \leq 0.004 \quad (13)$$

$$\sigma_{eff} = E_{FRCM}\varepsilon_{eff} \quad (14)$$

397 Eq. (13) limits the maximum strain to the lesser of the ultimate tensile strain of FRCM composite  $\varepsilon_{FRCM,u}$  and  
 398 0.004. Unfortunately, the guideline does not discuss evidence behind the 0.004 limit and/or the type of failure that  
 399 is intended to be prevented by imposing this limitation. However, it is worth noting that the ACI 440.2R guide  
 400 [41] imposes the same limitation for FRP composite strengthening systems to preclude the loss of aggregate  
 401 interlock or delamination of FRP from the substrate for completely wrapped and two- or three-sided wrapping  
 402 configurations.

403 Figure 9 plots  $v_{test}$  versus  $v_{pred}$  using Model 3 for the tests included in DB3, and Table 5 summarizes values of  
 404 AVG, STD and  $COV_1$ . For beams that failed by detachment of the strengthening system, most points (9 of 13)  
 405 fall below the  $v_{test}/v_{pred}=1.0$  line, i.e., unsafe results, and  $AVG=1.03$  (Table 5). For beams that did not show  
 406 detachment of the FRCM composite from the substrate, all points plot above the  $v_{test}/v_{pred}=1.0$  line with  $AVG=$   
 407 3.70. It is important to highlight that the six tests that comprise the no detachment subgroup are from a single  
 408 reference [18]. Regarding the accuracy of the model, results in Figure 9 and Table 5 show that it is highly affected  
 409 by the failure mode. The  $COV_1$  for beams with detachment is considerably lower (0.68) than that of beams with no  
 410 detachment (3.02).



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 412  
 413 Figure 9  $v_{test}$  versus  $v_{pred}$  for Model 3: a) DB3; b) DB3-Detachment; c) DB3-No detachment  
 414  
 415  
 416

Table 5  $V_{test}/V_{pred}$  Model 3 with DB3

Sample	#	AVG	STD	COV <sub>1</sub>
Detachment	13	1.03	0.68	0.68
No Detachment	6	3.70	1.36	3.02
Total	19	1.87	1.56	1.79

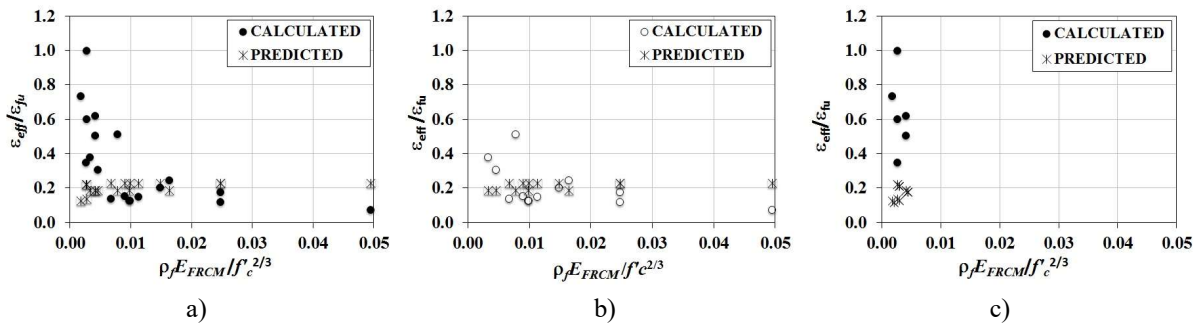
417  
418 A possible explanation of performance of the model could be related to the limitation of design strain imposed by  
419 the model. In fact, when Eq. (13) is applied to the 19 tests in DB3, the limiting value of 0.004 controls the value  
420 of  $\varepsilon_{eff}$  for each beam, i.e.  $\varepsilon_{FRCM,u}$  is always higher than the limit imposed by the model.

421 Rearranging Eq. (12), Eq. (15) can be used to determine the effective strain in the FRCM composite from the  
422 value of  $V_{test}$ :

$$\varepsilon_{eff} = \frac{V_{test}}{nA_f E_{FRCM} d} \quad (15)$$

423 Values of  $\varepsilon_{eff}/\varepsilon_{fu}$  are plotted against  $\rho_f E_{FRCM}/f_c^{2/3}$  for all tests in DB3 in Figure 10 where  $\varepsilon_{eff}$  is calculated using  
424 Eq. (15), and are shown as “calculated” in the graph. It is important to highlight that for this model,  $E_{FRCM}$  is used  
425 instead of  $E_f$ . Figure 10 also includes the strains used to compute  $V_{pred}$ , normalized by the ultimate strain of the  
426 FRCM composite, shown as “predicted” in the graph.

427



428  
429  
430 Figure 10 Normalized fiber strain  $\varepsilon_{eff}/\varepsilon_{fu}$  versus  $\rho_f E_{FRCM}/f_c^{2/3}$ : a) DB3; b) DB3-Detachment; c) DB3-No  
431 detachment  
432

433 Figure 10 shows that strains calculated by the model (predicted) are always lower than 25% of the ultimate strain  
434 of the composite. However, while these values appear to agree with the calculated  $\varepsilon_{eff}$  for larger values of  
435  $\rho_f E_{FRCM}/f_c^{2/3}$ , they do not agree for small values of  $\rho_f E_{FRCM}/f_c^{2/3}$ . The agreement between the calculated and  
436 predicted strains is clearer for beams that failed due to detachment of the FRCM system. All beams that failed by  
437 detachment have values of  $\rho_f E_{FRCM}/f_c^{2/3}$  larger than 0.003, while 83% of the remaining tests (i.e., tests that showed  
438 no detachment) present lower values. This suggests that  $\rho_f E_{FRCM}/f_c^{2/3}$  influences the failure mode.

### 439 3.3.2 Model 4: Ombres [20]

440 The model by Ombres [20], developed based on the experimental response of U-wrapped beams, computes  $V_{FRCM}$

441 by Eq. (16):

$$V_{FRCM} = k_e \varepsilon_{eff} E_{FRCM} \rho_f b d (\cot \alpha + \cot \theta) \sin \alpha \quad (16)$$

442 where  $k_e$  is an “effectiveness coefficient” that relates the strain in an FRP system to an FRCM system and is taken  
443 as 0.5, and the other variables were defined previously.

444 The effective strain  $\varepsilon_{eff}$  is computed based on the formulation adopted by the 2004 Italian CNR-DT 200 Guidelines  
445 [38] shown in Eq. (17) and (18):

$$\varepsilon_{eff} = \frac{f_{fdd}}{E_{FRCM}} \left[ 1 - \frac{1}{3} \frac{l_e \sin \alpha}{\min(0.9d; h_w)} \right] \quad (17)$$

$$f_{fdd} = \frac{0.24}{\gamma_{fd} \sqrt{\gamma_c}} \sqrt{\frac{E_{FRCM} k_b \sqrt{f_{ck} f_{ctm}}}{t_f}} \quad (18)$$

447 where  $f_{ck}$  is the concrete characteristic strength, and  $f_{ctm}$  is the mean value of concrete tensile strength computed  
448 as:

$$f_{ctm} = 0.30 f_{ck}^{2/3} \quad (19)$$

449 The partial safety factors,  $\gamma_{fd}$  and  $\gamma_c$ , are set to 1.0 in this analysis. The geometric coefficient  $k_b$  is calculated with  
450 Eq. (20):

$$k_b = \left[ \frac{2 - w_f/b}{1 + w_f/400} \right]^{0.5} \quad (20)$$

451 where  $b$  is equal to  $s_f$  for discontinuous strips or  $0.9d \sin(\theta + \alpha) / \sin \alpha$  for continuous configuration. The ratio  $w_f/b$   
452 should be larger than 0.33, otherwise the value of  $k_b$  with  $w_f/b$  equal to 0.33 shall be adopted. The optimal bond  
453 length,  $l_e$ , is defined as “the length, if exceeded, having no increase in the force transferred between concrete and  
454 FRP” [41]. Model 4 uses the expression in the 2004 Italian CNR-DT 200 Guidelines [38] for FRP systems to  
455 evaluate  $l_e$  and applies it to FRCM systems:

$$l_e = \left[ \frac{E_{FRCM} t_f}{2 f_{ctm}} \right] \quad (21)$$

456 It should be noted that the term  $l_e$  has not yet been clearly defined for the case of FRCM composites. Results have  
457 shown that debonding of the FRCM-concrete interface can occur within the composite itself at the fiber-matrix  
458 interface, as opposed to the composite-concrete interface with FRP [30]. In fact, for the case of some FRCM  
459 composites where debonding is associated with slippage of the fibers relative to the embedding matrix [31], the  
460 force transferred between the concrete and the FRCM composite has been shown to increase even after the stress  
461 transfer zone (STZ) is fully established because of friction (interlocking) between fibers and the matrix in the  
462 portion of the composite where the fibers have debonded [29]. Other work suggests that the concrete strength may  
463 not significantly influence the load-carrying capacity of the interface [42]. Therefore, the use of Eq. (21) for the  
464 case of FRCM composites may not be appropriate and requires further study.

465 In Figure 11,  $v_{test}$  is plotted against  $v_{pred}$  for Model 4. For beams that failed by detachment, most points fall close  
 466 to the line  $v_{test}/v_{pred}=1.0$  in Figure 11b. Figure 11c, on the other hand, shows that the model highly underestimates  
 467 the contribution of the FRCM system in the overall shear strength of beams with no detachment.  
 468 Table 6 presents the values of AVG, STD, and  $COV_1$  determined for Model 4 and the tests in DB3. For beams  
 469 that failed by composite detachment, the model predicts  $V_{FRCM}$  with good accuracy with  $AVG=1.14$  and  
 470  $COV_1=0.48$ . It is worth pointing out that five out of the 13 tests available are from [20] and therefore were used  
 471 to calibrate Model 4. For beams with no detachment, the model tends to highly underestimate the contribution of  
 472 the FRCM system, and the accuracy is relatively low. The poorer performance of the model for beams with no  
 473 detachment negatively affects the performance of the model when all 19 available tests are evaluated, as inferred  
 474 by the values of AVG and STD.

475 Table 6  $V_{test}/V_{pred}$  for Model 4 with DB3

Sample	#	AVG	STD	$COV_1$
Detachment	13	1.14	0.46	0.48
No Detachment	6	2.94	0.84	2.11
Total	19	1.71	1.34	1.25

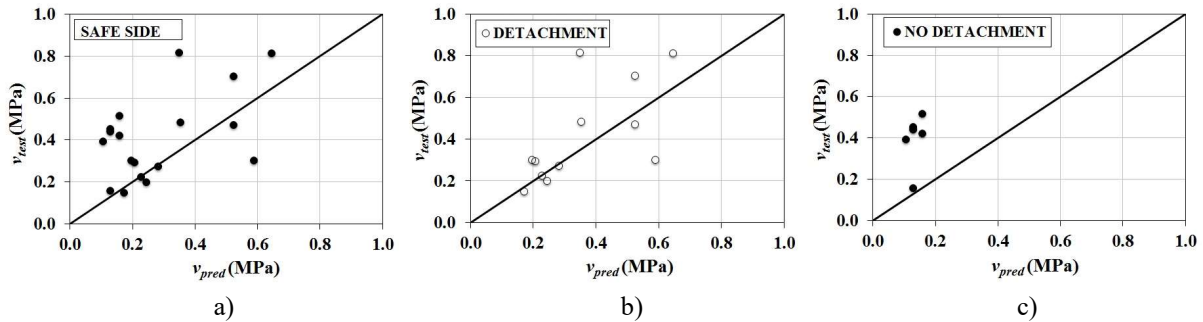


Figure 11  $V_{test}/V_{pred}$  ratios versus  $v$  for Model 4: a) DB3; b) DB3-Detachment; c) DB3-No Detachment

481 Rearranging Eq. (16), the effective strain can be computed from the value of  $V_{test}$  using Eq. (22):

$$\varepsilon_{eff} = \frac{V_{test}}{k_e E_{FRCM} \rho_f b d (\cot\alpha + \cot\theta) \sin\alpha} \quad (22)$$

482 In Figure 12,  $\varepsilon_{eff}/\varepsilon_{ju}$  ratios are plotted against  $\rho_f E_{FRCM}/f'_c{}^{2/3}$ , where  $\varepsilon_{long}$   
 483 is calculated using Eq. (22), and are shown as “calculated” in the graph. Figure 12 also includes the normalized  
 484 values of  $\varepsilon_{eff}$  computed using Eq. (17), shown as “predicted” in the graph. The behavior of Model 4 follows the  
 485 same trend as Model 3 discussed in Section 3.3.1, but for beams that failed by detachment, the values of strain  
 486 used by the model are always less than the 50% of the ultimate strain of the FRCM composite.

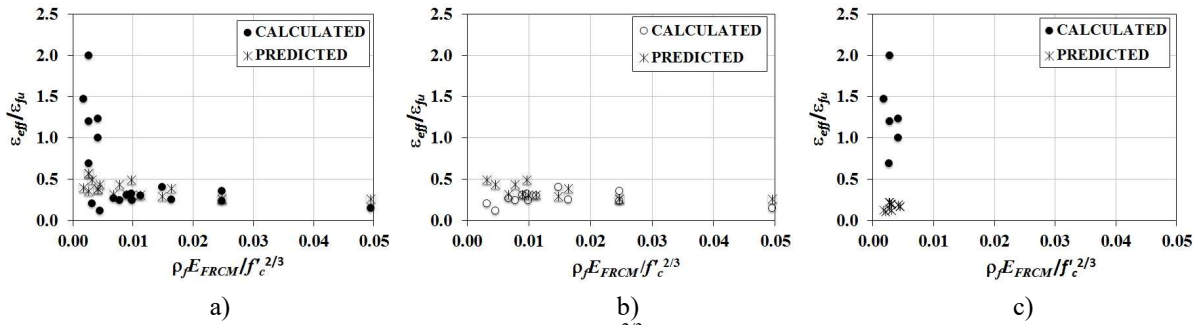


Figure 12 Normalized fiber strain in terms of  $\rho_f E_{FRCM} / f_c^{2/3}$ : a) Database 3; b) Detachment; c) No Detachment

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### 491 3.4 Comparison of the performance for Models 1, 2, 3, and 4

492 Table 7 summarizes values of AVG, STD, and  $COV_1$  determined for the four models studied. Since different  
493 subsets of the entire database were used in the assessment of each model, Table 7 includes the database and  
494 number of points considered for each analysis. As discussed in Section 3.1 the performance of Models 1 and 2  
495 can be compared using DB2, since specimens in DB2 are common to both models. The performance of Models  
496 1, 3, and 4 can be compared using DB3, since specimens in DB3 are common to all three models.

497 Although it was calibrated using a larger database, the AVG value obtained by Model 2 (1.35) is larger than the  
498 value obtained by Model 1 (1.12) when the common dataset DB2 is considered. The fact that Model 2 is only  
499 recommended for beams in which composite detachment is prevented limits its applicability.

500

Table 7  $V_{test}/V_{pred}$  for all models with different databases

DB	Model	Failure Mode	#	AVG	STD	$COV_1$
1	1	Detachment	33	0.80	0.75	0.86
	1	No detachment	36	1.12	0.71	0.72
	1	All	69	0.97	0.79	0.79
2	1	Detachment	36	1.12	0.71	0.72
	2			1.35	0.68	0.77
	3			1.03	0.68	0.68
3	1	Detachment	13	0.26	0.11	0.75
	3			1.03	0.68	0.68
	4			1.14	0.46	0.48
	1	No detachment	6	0.72	0.33	0.43
	3			3.70	1.36	3.02
	4			2.94	0.84	2.11
4	1	All	19	0.40	0.30	0.67
	3			1.87	1.56	1.79
	4			1.71	1.34	1.25

501

502 The model with the AVG value closest to 1.0 is Model 3 considering only beams that failed due to detachment  
503 (1.03). On the other hand, the largest AVG is also found for Model 3 (3.70) for beams that did not show  
504 detachment. Model 1 tends to highly overestimate the contribution of the FRCM system for beams that showed  
505 detachment with AVG values as low as 0.26 (DB3). Considering tests with both failure modes, Model 3 has an

506 AVG of 1.87, which is somewhat misleading since its performance is highly affected by failure mode.  
507 Models based on FRCM composite properties (Models 3 and 4) have AVG values close to 1.0 for beams that  
508 showed detachment. However, these models were not capable of accurately predicting the FRCM composite shear  
509 contribution for beams that did not show detachment with large values of AVG and  $COV_1$ .  
510 In general, although based on limited experimental evidence, Model 1 presents a more consistent behavior in  
511 terms of  $COV_1$  for both failure modes within all the databases. However, further work is needed to validate each  
512 model presented as more data become available.

513

#### 514 4. CONCLUSIONS

515 In this study, experimental results from 15 papers on shear strengthening of RC beams using externally bonded  
516 FRCM composites were collected. As result, a database that includes 89 tests was compiled, and the influence of  
517 geometrical and mechanical properties of the beams and the strengthening system was assessed. The database was  
518 also used to evaluate the performance of four models for the prediction of the contribution of the shear strength  
519 of FRCM composites to the overall strength of RC beams. The main conclusions drawn from this study are  
520 summarized as follows:

- 521 • The experimental evidence shows that FRCM composites are able to increase the shear strength of RC beams.  
522 For the beams included in the database, an increase of 3% to 195% was reported, with an average of 55%. In  
523 addition, the FRCM composite can modify the type of failure from shear to a flexural mode.
- 524 • The effectiveness of the FRCM system appears to be related to the compressive strength of the matrix, as  
525 lower values of  $V_{FRCM}/V_{CON}$  are usually found for matrixes with higher values of matrix compressive strength.  
526 The influence appears to be related to the compressive strength of the substrate, with larger values of  
527  $V_{FRCM}/V_{CON}$  reached when the compressive strengths of the matrix and the substrate are similar.
- 528 • As for FRP composites, a possible interaction between the internal transverse steel reinforcement and the  
529 FRCM system has been observed. As reported by some researchers, the presence of the FRCM composite  
530 can limit the strain in internal stirrups and prevent them from achieving their maximum possible contribution  
531 (based on yielding), resulting in lower values of  $V_{FRCM}/V_{CON}$ . Based on the experimental tests collected in this  
532 paper, this effect appears to be more pronounced for higher values of the ratio  $\rho_w E_s / \rho_f E_f$ .
- 533 • For fully wrapped beams, the failure mode has been associated with fracture of the fibers. For side bonded  
534 and U-wrapped beams, detachment of the FRCM jackets (with or without concrete attached) has been  
535 reported, being the most common failure mode for U-wrapped configurations. However, failure without

536 detachment has also been witnessed together with diagonal cracking, slippage of the vertical fibers through  
537 the mortar, and/or partial fiber rupture.

- 538 • Although Model 1 overestimates the additional shear strength provided by the FRCM composite, it presents  
539 the more consistent behavior regarding values of  $COV_1$  when compared with the other models. Models based  
540 on FRCM composite properties (Model 3 and 4) perform well for beams that failed by detachment, but they  
541 do not perform well for beams with no detachment.
- 542 • Although it was calibrated using a larger database, the AVG value obtained by Model 2 (1.35) is larger than  
543 the value obtained by Model 1 (1.12) considering a common dataset. In addition, the fact that Model 2 is only  
544 recommended for beams in which composite detachment is prevented limits its applicability.
- 545 • The use of the properties of the FRCM composite instead of the fiber mechanical characteristics in Models 3  
546 and 4 does not result in a significant increase in accuracy of the models, measured in terms of  $COV_1$ . In fact,  
547 a simple formulation such as the one proposed by Model 1, based on fiber properties, is more accurate for  
548 beams with or without composite detachment.
- 549 • According to the available experimental results, strengthening solutions with values of  $\rho_f E_f / f_c^{2/3}$  lower than  
550 0.02 might avoid the onset of composite detachment. It was also observed that having less stiff solutions, i.e.,  
551 lower values of  $\rho_f E_f$  or  $\rho_f E_{FRCM}$ , results in a better exploitation of the FRCM system.

552 The above conclusions will need to be validated when more experimental data become available. It is also hoped  
553 that the evaluation of the database and distribution of data carried out in this paper will help researchers to plan  
554 future experimental tests that focus on variables with scarce data, such as strains in internal transverse shear  
555 reinforcement, the influence of the ratio  $f_{cm}/f_c$ , the study of different type of fibers, and how the use of anchors  
556 help mitigate detachment and other forms of FRCM composite debonding. In addition, the inclusion of variable  
557 shear crack angles in the design models needs to be studied to evaluate their influence and potentiality improve  
558 the available models. Furthermore, the interaction between the internal and external shear reinforcement requires  
559 special attention in the development of future design models aimed to compute the final total shear capacity of a  
560 strengthened element, as the simple addition of concrete, steel, and FRCM composite contributions might not  
561 provide accurate results.

562

## 563 **ACKNOWLEDGEMENTS**

564 This work was supported by the European Commission (Contract number MC-ITN-2013-607851). The first and  
565 second authors would like to acknowledge the technical and economical support from the European Network for

566 Durable Reinforcement and Rehabilitation Solutions (*endure*), a Marie Skłodowska Curie Initial Training  
567 Network.

568

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## APPENDIX A

Table A1 Experimental database

Ref.	Name	Shape	Geometry			Concrete*	Int. Reinf.					FRCM Composite										Results	
			$b_w$ [mm]	$d$ [mm]	$a/d$	$f_c$ [MPa]	$\rho_{long}$	$\rho_w$	SC	Fiber	Anchors	$s_f$ [mm]	$w_f$ [mm]	$E_f$ [GPa]	$f_f$ [MPa]	$n$	$\rho_f$	$f'_{cm}$ [MPa]	$E_{FRCM}$ [GPa]	$\rho_{cm}$	Failure mode	$V_{FRCM}$ [kN]	$V_{FRCM}/V_{CON}$
[9]	M2	R	150	272	2.85	25.3	0.015	0.0014	W	C	No	1	1	225	3350	2	0.0013	30.6	NR	0.070	F	63.7	1.09
	M2-s	R	150	272	2.85	25.3	0.015	0.0014	W	C	No	1	1	225	3350	2	0.0012	30.6	NR	0.070	F	60.6	1.04
	M1	R	150	272	2.85	25.3	0.015	0.0014	W	C	No	1	1	225	3350	1	0.0006	30.6	NR	0.047	S	41.8	0.72
[10]	R2	R	150	256	3.91	23.2	0.032	0.0000	W	G	No	1	1	75	574	2	0.0015	77.2	NR	0.080	S	25.5	0.44
	R3	R	150	256	3.91	23.2	0.032	0.0000	W	G	No	1	1	75	574	3	0.0022	77.2	NR	0.053	S	43.5	0.74
[12]	PB-1/1	T	120	372	2.69	25.5	0.042	0.0042	U	G	No	1	1	75	574	2	0.0018	82.8	NR	0.100	S	44.7	0.19
	PB-1/2	T	120	372	2.69	26.3	0.042	0.0042	U	G	No	1	1	75	574	4	0.0037	85.3	NR	0.167	S	41.5	0.18
	PB-1/3	T	120	372	2.69	28.6	0.042	0.0042	U	G	No	1	1	75	574	6	0.0055	79.3	NR	0.233	S	46.8	0.19
	PB-2/1	T	120	372	2.69	27.1	0.042	0.0042	U	G	Yes	1	1	75	574	2	0.0018	70.6	NR	0.100	S	51.3	0.21
	PB-2/2	T	120	372	2.69	25.6	0.042	0.0042	U	G	Yes	1	1	75	574	4	0.0037	86.7	NR	0.167	S	67.4	0.29
	PB-2/3	T	120	372	2.69	28.7	0.042	0.0042	U	G	Yes	1	1	75	574	6	0.0055	75.4	NR	0.233	S	72.4	0.29
	PB-3/1	T	120	372	2.69	28.0	0.042	0.0042	U	G	Yes	1	1	75	574	3	0.0027	72.0	NR	0.133	S	34.1	0.14
	PB-3/2	T	120	372	2.69	34.0	0.042	0.0042	U	G	Yes	1	1	75	574	3	0.0027	79.1	NR	0.133	S	42.1	0.16
PB-3/3	T	120	372	2.69	32.0	0.042	0.0042	U	G	Yes	1	1	75	574	4	0.0037	63.3	NR	0.167	S	56.9	0.22	
[13]	C40s0-M2-G2a	R	180	419	2.98	46.2	0.032	0.0000	SB	C	No	1	1	253	3800	1	0.0002	45.0	NR	0.222	S	59.9	0.96
	C40s0-M2-G2b	R	180	419	2.98	46.2	0.032	0.0000	SB	C	No	1	1	253	3800	1	0.0002	45.0	NR	0.222	S	58.4	0.93
	C40s0-M3-G2	R	180	419	2.98	46.2	0.032	0.0000	SB	C	No	1	1	201	3800	1	0.0002	77.0	NR	0.222	S	55.0	0.88
	C40s0-M2-G1	R	180	419	2.98	46.2	0.032	0.0000	SB	C	No	1	1	253	3800	1	0.0002	45.0	NR	0.222	S	41.5	0.66
	C40s0-M2-G2	R	180	419	2.98	46.2	0.032	0.0000	SB	C	No	1	1	253	3800	1	0.0002	45.0	NR	0.222	S	63.4	1.01
	C40s0-M2-G3	R	180	419	2.98	46.2	0.032	0.0000	SB	C	No	1	1	253	3800	1	0.0002	45.0	NR	0.222	S	40.7	0.65
	C40s0-M1-G3	R	180	419	2.98	46.2	0.032	0.0000	SB	C	No	1	1	262	2950	1	0.0002	22.0	NR	0.222	S	27.5	0.44
[1]	BS2	R	150	159	2.52	20.0	0.013	0.0000	SB	B	No	1	1	31.9	623	2	0.0017	23.9	NR	0.080	S	10.9	0.36
	BS3	R	150	159	2.52	20.0	0.013	0.0000	SB	B	No	1	1	31.9	623	2	0.0012	23.9	NR	0.080	S	11.3	0.37
	BS4	R	150	159	2.52	20.0	0.013	0.0000	SB	B	No	1	1	31.9	623	4	0.0034	23.9	NR	0.133	S	14.0	0.46
	BS5	R	150	159	2.52	20.0	0.013	0.0000	SB	B	No	1	1	31.9	623	4	0.0024	23.9	NR	0.133	S	15.8	0.52
	BS6	R	150	159	2.52	20.0	0.013	0.0000	SB	B	No	1	1	31.9	623	2	0.0017	56.4	NR	0.080	S	11.3	0.37
	BS7	R	150	159	2.52	20.0	0.013	0.0000	SB	B	No	1	1	31.9	623	2	0.0012	56.4	NR	0.080	S	11.3	0.37
	BS8	R	150	159	2.52	20.0	0.013	0.0000	SB	B	No	1	1	31.9	623	4	0.0034	56.4	NR	0.133	S	17.7	0.58
	BS9	R	150	159	2.52	20.0	0.013	0.0000	SB	B	No	1	1	31.9	623	4	0.0024	56.4	NR	0.133	S	26.6	0.88
	[14]	R30-C-UJ-HI-TRC(5)	R	120	204	3.18	25.6	0.026	0.0000	U	G	No	1	1	74	1102	1	0.0012	42.0	2.72	0.083	F	30.3
R30-S-SB-P-TRC(10)		R	120	204	3.18	25.6	0.026	0.0000	SB	G	No	120	100	74	1102	1	0.0010	42.0	2.72	0.139	F	28.3	0.36
R30-S-SB-P-TRC(5)		R	120	204	3.18	25.6	0.026	0.0000	SB	G	No	120	100	74	1102	1	0.0010	42.0	2.72	0.069	S	25.3	0.32
R30-S-UJ-HI-TRC(5)		R	120	204	3.18	25.6	0.026	0.0000	U	G	No	200	40	74	1102	1	0.0002	42.0	2.72	0.017	S	5.8	0.07
R40-S-UJ-HI-TRC(5)		R	120	204	3.18	35.2	0.026	0.0000	U	G	No	200	100	74	1102	1	0.0006	42.0	2.72	0.042	S	11.0	0.10
R40-C-UJ-HI-TRC(2)		R	120	204	3.18	35.2	0.026	0.0000	U	G	No	1	1	74	1102	1	0.0012	42.0	2.72	0.033	S	3.0	0.03
[15]	SB-GT	R	150	308	3.25	37.5	0.021	0.0000	SB	G	No	1	1	75	2300	1	0.0006	58.0	NR	0.093	S	11.4	0.18

Ref.	Name	Shape	Geometry			Concrete*		Int. Reinf.				FRCM Composite								Results			
			$b_w$ [mm]	$d$ [mm]	$a/d$	$f'_c$ [MPa]	$\rho_{long}$	$\rho_w$	SC	Fiber	Anchors	$s_f$ [mm]	$w_f$ [mm]	$E_f$ [GPa]	$f_f$ [MPa]	$n$	$\rho_f$	$f'_{cm}$ [MPa]	$E_{FRCM}$ [GPa]	$\rho_{cm}$	Failure mode	$V_{FRCM}$ [kN]	$V_{FRCM}/V_{CON}$
	UW-GT	R	150	308	3.25	37.5	0.021	0.0000	U	G	No	1	1	75	2300	1	0.0006	58.0	NR	0.093	S	28.4	0.46
	SB-CT1	R	150	308	3.25	37.5	0.021	0.0000	SB	C	No	1	1	230	3800	1	0.0005	58.0	NR	0.093	S	16.0	0.26
	UW-CT1	R	150	308	3.25	37.5	0.021	0.0000	U	C	No	1	1	230	3800	1	0.0005	58.0	NR	0.093	S	14.2	0.23
	SB-CT2	R	150	308	3.25	37.5	0.021	0.0000	SB	C	No	1	1	230	3800	1	0.0012	58.0	NR	0.093	S	61.0	0.99
	UW-CT2	R	150	308	3.25	37.5	0.021	0.0000	U	C	No	1	1	230	3800	1	0.0012	58.0	NR	0.093	S	65.0	1.05
[16]	Beam 4	R	150	310	2.90	41.6	0.030	0.0021	U	G	No	275	200	75	2300	1	0.0010	40.0	NR	0.058	S	35.5	0.32
	Beam 5	R	150	310	2.90	41.6	0.030	0.0021	U	G	Yes	275	200	75	2300	1	0.0010	40.0	NR	0.058	S	38.5	0.35
	L1	T	150	320	2.50	16.7	0.016	0.0000	U	C	No	1	1	225	3375	1	0.0006	21.8	NR	0.053	S	9.6	0.17
	L2	T	150	320	2.50	18.0	0.016	0.0000	U	C	No	1	1	225	3375	2	0.0013	21.8	NR	0.080	S	11.4	0.19
	H1	T	150	320	2.50	19.4	0.016	0.0000	U	C	No	1	1	225	3375	1	0.0013	21.8	NR	0.053	S	19.9	0.32
	H2	T	150	320	2.50	19.2	0.016	0.0000	U	C	No	1	1	225	3375	2	0.0026	21.8	NR	0.080	S	33.1	0.54
[17]	L2A15	T	150	320	2.50	20.1	0.016	0.0000	U	C	Yes	1	1	225	3375	2	0.0013	21.8	NR	0.080	S	51.8	0.83
	L2A15ha	T	150	320	2.50	19.2	0.016	0.0000	U	C	Yes	1	1	225	3375	2	0.0013	21.8	NR	0.080	S	55.6	0.91
	L2A10	T	150	320	2.50	10.1	0.016	0.0000	U	C	Yes	1	1	225	3375	2	0.0013	21.8	NR	0.080	S	84.3	1.87
	H1A15	T	150	320	2.50	10.7	0.016	0.0000	U	C	Yes	1	1	225	3375	1	0.0013	21.8	NR	0.053	S	51.9	1.12
	H2A15	T	150	320	2.50	11.1	0.016	0.0000	U	C	Yes	1	1	225	3375	2	0.0026	21.8	NR	0.080	S	48.0	1.01
	H2A10	T	150	320	2.50	20.8	0.016	0.0000	U	C	Yes	1	1	225	3375	2	0.0026	21.8	NR	0.080	S	45.6	0.72
	V-BR3-01	R	300	254	2.76	28.0	0.008	0.0007	U	B	No	1	1	95	2990	1	0.0004	24.6	48	0.067	S	29.9	0.44
	V-CXM25-01	R	300	254	2.76	28.0	0.008	0.0007	U	C	No	1	1	240	4320	1	0.0003	25.0	80	0.067	S	34.3	0.50
	V-CXM25-02	R	300	254	2.76	28.3	0.008	0.0007	U	C	No	1	1	240	4320	1	0.0003	25.0	80	0.067	S	11.9	0.17
[18]	V-PXM750-01	R	300	254	2.76	28.3	0.008	0.0007	U	PBO	No	1	1	270	5800	1	0.0003	30.0	128	0.067	S	31.9	0.46
	V-PXM750-02	R	300	254	2.76	28.3	0.008	0.0007	U	PBO	No	1	1	270	5800	1	0.0003	30.0	128	0.067	S	39.2	0.57
	V-GPHDM-02	R	300	254	2.76	28.3	0.008	0.0007	U	G	No	1	1	90	2610	1	0.0003	35.4	90	0.067	S	33.4	0.48
	W600-L1	R	150	270	2.22	28.0	0.015	0.0000	SB	C	No	1	1	240	4300	1	0.0014	45.0	160	0.067	S	19.0	0.36
	W600-L2	R	150	270	2.22	28.0	0.015	0.0000	SB	C	No	1	1	240	4300	2	0.0029	45.0	160	0.100	S	23.5	0.45
	W50-N4	R	150	270	2.22	28.0	0.015	0.0000	U	C	No	183	50	240	4300	1	0.0004	45.0	160	0.018	S	6.0	0.11
[19]	W50-N5	R	150	270	2.22	28.0	0.015	0.0000	U	C	No	138	50	240	4300	1	0.0005	45.0	160	0.024	S	9.0	0.17
	W50-N6	R	150	270	2.22	28.0	0.015	0.0000	U	C	No	110	50	240	4300	1	0.0006	45.0	160	0.030	S	11.0	0.21
	W100-N3	R	150	270	2.22	28.0	0.015	0.0000	U	C	No	250	100	240	4300	1	0.0006	45.0	160	0.027	S	8.0	0.15
	W100-N4	R	150	270	2.22	28.0	0.015	0.0000	U	C	No	167	100	240	4300	1	0.0009	45.0	160	0.040	S	19.5	0.37
	W600-N1	R	150	270	2.22	28.0	0.015	0.0000	U	C	No	1	1	240	4300	1	0.0014	45.0	160	0.067	S	28.5	0.54
	TRA1	R	150	225	3.00	30.8	0.019	0.0023	U	PBO	No	1	1	270	5800	1	0.0006	30.4	128	0.107	F	19.0	0.25
	TRA2	R	150	225	3.00	30.8	0.019	0.0023	U	PBO	No	260	150	270	5800	1	0.0004	30.4	128	0.062	S	9.9	0.13
[20]	TRB1	R	150	225	2.78	45.0	0.028	0.0032	U	PBO	No	1	1	270	5800	1	0.0006	30.4	128	0.107	F	34.2	0.32
	TRB2	R	150	225	2.78	29.2	0.028	0.0032	U	PBO	No	1	1	270	5800	2	0.0012	30.4	128	0.160	S	27.4	0.40
	TRB3	R	150	225	2.78	29.2	0.028	0.0032	U	PBO	No	210	100	270	5800	2	0.0006	30.4	128	0.076	S	27.5	0.40
	TRB4	R	150	225	2.78	38.3	0.028	0.0032	U	PBO	No	210	100	270	5800	1	0.0003	30.4	128	0.051	S	10.2	0.11
	TRB5	R	150	225	2.78	38.3	0.028	0.0032	U	PBO	No	210	100	270	5800	3	0.0009	30.4	128	0.102	S	10.2	0.11
[21]	SB_M1	R	102	177	2.60	21.6	0.022	0.0000	SB	C	No	1	1	225	3800	1	0.0019	31.1	NR	0.078	S	2.7	0.09

Ref.	Name	Shape	Geometry			Concrete*		Int. Reinf.			FRCM Composite								Results				
			$b_w$ [mm]	$d$ [mm]	$a/d$	$f'_c$ [MPa]	$\rho_{long}$	$\rho_w$	SC	Fiber	Anchors	$s_f$ [mm]	$w_f$ [mm]	$E_f$ [GPa]	$f_f$ [MPa]	$n$	$\rho_f$	$f'_{cm}$ [MPa]	$E_{FRCM}$ [GPa]	$\rho_{cm}$	Failure mode	$V_{FRCM}$ [kN]	$V_{FRCM}/V_{CON}$
	SB_M2	R	102	177	2.60	22.6	0.022	0.0000	SB	C	No	1	1	225	3800	2	0.0037	28.2	NR	0.118	S	15.1	0.51
	SB_M3	R	102	177	2.60	22.6	0.022	0.0000	SB	C	No	1	1	225	3800	3	0.0056	26.9	NR	0.157	S	34.0	1.14
	UW_M1	R	102	177	2.60	23.8	0.022	0.0000	U	C	No	1	1	225	3800	1	0.0019	31.1	NR	0.078	S	21.1	0.71
	UW_M2	R	102	177	2.60	23.8	0.022	0.0000	U	C	No	1	1	225	3800	2	0.0037	31.1	NR	0.118	S	39.1	1.32
	UW_M3	R	102	177	2.60	22.6	0.022	0.0000	U	C	No	1	1	225	3800	3	0.0056	26.9	NR	0.157	S	57.8	1.95
	FW_M1	R	102	177	2.60	21.6	0.022	0.0000	W	C	No	1	1	225	3800	1	0.0019	31.1	NR	0.078	S	32.7	1.10
	FW_M2	R	102	177	2.60	21.6	0.022	0.0000	W	C	No	1	1	225	3800	2	0.0037	28.2	NR	0.118	F	45.4	1.53
[22]	B1	R	150	204	4.90	42.9	0.051	0.0013	W	PBO	No	200	100	270	5270	1	0.0003	29.0	NR	0.040	S	70.1	1.19
	S0-FRCM-1	R	150	250	3.00	36.0	0.050	0.0000	SB	C	No	1	1	230	3800	1	0.0004	74.0	NR	0.160	S	66.8	1.11
	S0-FRCM-2	R	150	250	3.00	36.0	0.050	0.0000	SB	C	No	1	1	230	3800	2	0.0008	74.0	NR	0.240	S	87.5	1.46
[23]	S1-FRCM-1	R	150	250	3.00	36.0	0.050	0.0025	SB	C	No	1	1	230	3800	1	0.0004	74.0	NR	0.160	S	68.4	0.64
	S1-FRCM-2	R	150	250	3.00	36.0	0.050	0.0025	SB	C	No	1	1	230	3800	2	0.0008	74.0	NR	0.240	S	72.1	0.67
	S2-FRCM-1	R	150	250	3.00	36.0	0.050	0.0050	SB	C	No	1	1	230	3800	1	0.0004	74.0	NR	0.160	S	67.7	0.51
	S2-FRCM-2	R	150	250	3.00	36.0	0.050	0.0050	SB	C	No	1	1	230	3800	2	0.0008	74.0	NR	0.240	S	73.6	0.55

\*For references reporting cube compressive strength ( $f'_{c,cube}$ ),  $f'_c$  was computed according to [43]

Fiber: C=Carbon, G=Glass, B=Basalt. SC: Strengthening configuration, see Table 1. Failure mode: F=Flexure, S=Shear, see Table 1. NR=Not reported.

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